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1 A macro-element strategy based upon spectral finite elements and mortar
2 elements for transient wave propagation modeling. Application to
3 ultrasonic testing of stratified composite materials

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7 **Abstract**

8 Proposing efficient numerical modeling tools for high-frequency wave propagation in realistic configurations,
9 such as the one appearing in ultrasonic testing experiments, is a major challenge, especially in the per-
10 spective of inversion loops or parametric studies. We propose a numerical methodology addressing this
11 challenge, and based upon the combination of the spectral finite element method and the mortar element
12 method. From a prior decomposition of the scene of interest into “macro-elements” we show how one can
13 improve the performances of the standard finite element procedures in terms of memory footprint and com-
14 putational load. Additionally, using this decomposition, we are able to efficiently reconstruct important
15 modeling features on-the-fly, such as orientations of anisotropic materials or splitting directions of perfectly
16 matched layers formulations, altogether in a robust and efficient manner. We believe that this strategy is
17 particularly suitable for parametric studies and sensitivity analysis. We illustrate our strategy by simulating
18 the propagation of an ultrasonic wave into an immersed and curved anisotropic stratified 3D specimen flawed
19 with an internal circular delamination of varying size, thus showing the efficiency and the robustness of our
approach.

20 *Keywords:* finite element methods; time integration; explicit; acoustics; structures

21 **1. Introduction**

22 Throughout decades of their intensive use, Non Destructive Testing (NDT) techniques have become a
23 major asset in numerous advanced industrial fields such as nuclear energy industry, petrochemical industry
24 or aeronautics. Traditional Ultrasonic Testing (UT) methods [1] based upon the propagation of a high-
25 frequency bulk wave within materials or manufactured products keep a major place among existing NDT
26 techniques. Concurrently, the role of UT modeling [2] has been continuously increasing [3] as a support of
27 the experimental data analysis and post-processing [4, 5, 6], the analysis and design of transducers [7, 8, 9],
28 or the evaluation of UT processes [10, 11]. These examples of application often require exploring the output
29 response of a so-called forward solver by varying the input data. This sensitivity analysis typically entails
30 a large number of direct simulations that are generally embedded within an inversion algorithm. Hence,
31 proposing useful modeling tools should be performed following a primary goals:

32 (G1) The forward solver should be able to efficiently handle parametric variations of the configuration of
33 interest.

34 In the scope of this communication, the presented work is driven by a second major goal, which results
35 from our wish to make the modeling tools accessible by a non-specialist end-user, e.g. through a commercial
36 software such as the CIVA platform [12]:

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37 (G2) No specific hardware architecture should be required from the end-user in order to support the simu-
38 lation’s runs.

39 In the context of high-frequency wave propagation one can consider, without being fully comprehensive, at
40 least two principal families of modeling strategies. First, the class of asymptotic methods [13, 14] relying
41 on an asymptotic expansion of the wave field w.r.t. the frequency is widely spread in the NDT [15, 16] and
42 seismic waves [17, 18] modeling communities. This type of methods provides an efficient and meaningful way
43 to represent the propagation of waves over large distances. However, due to the truncation of the asymptotic
44 development, they fail to represent various – and potentially important – phenomena such as diffraction by
45 small flaws, or reflections at critical angles. Second, a full-wave modeling of the propagation phenomena
46 can be obtained using numerical methods, such as finite element methods [19, 20], finite difference methods
47 [21] or boundary element methods [22]. Nevertheless, as they are based upon discretization steps related
48 to the wavelength scale, they often require dedicated solvers based upon specific hardware architectures
49 [23, 24, 25, 26] in order to achieve sufficient performances.

50 In our work, we propose a UT modeling numerical tool based upon a “macro-element” strategy that
51 satisfies both our primary goals (G1) and (G2). This strategy relies on the assumption that the geometry
52 associated to the UT configuration can be decomposed into sub-domains. Each sub-domain is defined
53 as the transformation of a reference macro-element, the unit cube in 3D, and is assigned to a unique
54 formulation (e.g. acoustics, elastodynamics or absorbing layers, with potentially inhomogeneous material
55 properties). In order to provide an efficient fully discrete propagator, each macro-element is sub-discretized
56 – depending on the estimated wavelengths of interest – using the spectral finite element method [19, 20, 27].
57 The communication between the various formulations assigned to the macro-elements is handled using the
58 mortar element method [28, 29, 30]. By making the most of this decomposition we are able to increase
59 the performances of standard finite element procedures and to incorporate efficiently important modeling
60 features.

61 The outline of the paper is as follows. In Section 2 we recall the main components of the discrete
62 propagators built upon spectral finite elements for each formulation of interest. This enables us to detail
63 in Section 3 the macro-element strategy and how it addresses goals (G1) and (G2). In Section 4 we show
64 how the mortar element method can be used in practice to efficiently connect neighboring macro-elements
65 with potentially different formulations. In Section 5, we illustrate our strategy in the context of the UT
66 of curved Carbon Fiber Reinforced Polymer (CFRP) composite structures in a 3D setting. We finally give
67 some conclusions and perspectives in Section 6.

68 2. Discrete propagators based upon spectral finite elements

69 We consider the propagators for the fluid, solid and corresponding Perfectly Matched absorbing Layers
70 (PMLs). They form the minimal set of formulations that one needs to handle in order to address most
71 of the UT configurations. In this section, we recall the main components of the finite element method
72 employed to propose discrete propagators for every formulations, in order to clearly list the benefits brought
73 by the macro-element strategy in the next section. For further details on finite elements, we invite readers
74 to refer to [19, 20, 31, 32] for a comprehensive presentation in the case of acoustic and elastodynamics, and
75 to [33, 34, 35] for the PML formulations.

76 2.1. Fluid, solid and PML formulations

77 In the following, we consider $\Omega \subset \mathbb{R}^d$, with $d = 2$ or $d = 3$, a bounded computational domain bearing one
78 of the possible formulations. We represent by $\partial\Omega$ its boundary and by \underline{n} the corresponding exterior normal
79 vector field. We denote the coordinates in Ω by $\underline{x} = (x_1 \cdots x_d)^\top \in \mathbb{R}^d$, and by

$$80 \quad \nabla_{\underline{x}} u = \left(\frac{\partial u}{\partial x_i} \right)_{i=1}^d, \quad \nabla_{\underline{x}} v = \left(\frac{\partial v_i}{\partial x_j} \right)_{i,j=1}^d,$$

81 the gradient of a scalar and vector field respectively. Accordingly, the divergence operators of vector and
 82 tensor fields are denoted by

$$83 \quad \nabla_{\underline{x}} \cdot \underline{v} = \sum_{i=1}^d \frac{\partial v_i}{\partial x_i}, \quad \nabla_{\underline{x}} \cdot \underline{w} = \left(\sum_{j=1}^d \frac{\partial w_{ij}}{\partial x_j} \right)_{i=1}^d,$$

84 so that the scalar Laplacian operator simply reads $\Delta_{\underline{x}} u = \nabla_{\underline{x}} \cdot (\nabla_{\underline{x}} u)$. In the case of a fluid domain, the
 85 material is characterized by its mass density ρ , assumed constant for simplicity, and its sound velocity c .
 86 The wave equation satisfied by the acoustic pressure p reads

$$87 \quad \frac{1}{\rho c^2} \frac{\partial^2 p}{\partial t^2} - \frac{1}{\rho} \Delta_{\underline{x}} p = 0 \text{ in } \Omega, \quad \nabla_{\underline{x}} p \cdot \underline{n} = 0 \text{ on } \partial\Omega, \quad (1)$$

88 along with given initial conditions. The weak formulation associated to (1) consists in finding $p \in V = H^1(\Omega)$
 89 for any time $t > 0$, such that, for any test function $p^* \in V$,

$$90 \quad \frac{d^2}{dt^2} m(p, p^*) + k(p, p^*) = 0, \quad (2)$$

91 where $m(\cdot, \cdot)$ and $k(\cdot, \cdot)$ are two bilinear forms defined by

$$92 \quad m(p, p^*) = \int_{\Omega} \frac{1}{\rho c^2} p p^* d\Omega, \quad k(p, p^*) = \int_{\Omega} \frac{1}{\rho} \nabla_{\underline{x}} p \cdot \nabla_{\underline{x}} p^* d\Omega. \quad (3)$$

93 In the case of a solid domain, the displacement field \underline{y} satisfies the following field equation

$$94 \quad \rho \frac{\partial^2 \underline{y}}{\partial t^2} - \nabla_{\underline{x}} \cdot \underline{\sigma} = 0 \text{ in } \Omega, \quad \underline{\sigma} \cdot \underline{n} = 0 \text{ on } \partial\Omega, \quad (4)$$

95 with ρ being the mass density, and $\underline{\sigma}$ the stress tensor field. We complete (4) with relevant initial conditions.
 96 The linearized Green-Lagrange tensor and the stress tensor are linked through a linear constitutive law

$$97 \quad \underline{\sigma} = \mathcal{C} \underline{\varepsilon}(\underline{y}), \quad \underline{\varepsilon}(\underline{y}) = \frac{1}{2} (\nabla_{\underline{x}} \underline{y} + \nabla_{\underline{x}} \underline{y}^T).$$

98 We assume that the fourth order tensor \mathcal{C} can be decomposed into a constant tensor \mathcal{C}^* defined in a local
 99 orthonormal basis $\{\underline{e}_i\}_{i=1}^d$,

$$100 \quad \mathcal{C}(\underline{x}) = \sum_{i,j,k,l=1}^d C_{ijkl}^* \underline{e}_i(\underline{x}) \otimes \underline{e}_j(\underline{x}) \otimes \underline{e}_k(\underline{x}) \otimes \underline{e}_l(\underline{x}), \quad \forall \underline{x} \in \Omega, \quad (5)$$

101 and so that \mathcal{C} satisfies the standard symmetry and positivity conditions. The corresponding weak formulation
 102 is similar to (2) where, for any test function $\underline{y}^* \in V = [H^1(\Omega)]^d$, the bilinear forms are

$$103 \quad m(\underline{y}, \underline{y}^*) = \int_{\Omega} \rho \underline{y} \cdot \underline{y}^* d\Omega, \quad k(\underline{y}, \underline{y}^*) = \int_{\Omega} \mathcal{C} \underline{\varepsilon}(\underline{y}) : \underline{\varepsilon}(\underline{y}^*) d\Omega. \quad (6)$$

104 In addition to these natural formulations, we are interested in formulations addressing the challenge of
 105 numerically representing infinite propagation areas. Building such formations is an active field of research
 106 that goes beyond the scope of this paper. Hence, without going into details, we use as is a class of absorbing
 107 layers referred to as Perfectly Matched Layers (PMLs), and more specifically the formulation proposed in
 108 [35]. In the acoustic case, it is a first-order in time formulation based upon split pressure variables $\{p_i\}_{i=1}^d$

109 and a velocity variable \underline{v} , satisfying

$$110 \quad \begin{cases} p = \sum_{i=1}^d p_i, \\ \frac{1}{\rho c^2} \frac{\partial p_i}{\partial t} + \nabla_{\underline{x}} \cdot ((\underline{s}_i \otimes \underline{s}_i) \underline{v}) + \tau_i \frac{1}{\rho c^2} p_i = 0, \quad i = 1, \dots, d, \\ \rho \frac{\partial \underline{v}}{\partial t} + \sum_{i=1}^d \nabla_{\underline{x}} p_i + \sum_{i=1}^d \tau_i (\underline{s}_i \otimes \underline{s}_i) \underline{v} = 0. \end{cases} \quad (7)$$

111 The formulation (7) is completed with relevant initial conditions and boundary conditions. In (7), $\{\underline{s}_i\}_{i=1}^d$
 112 are the splitting directions, and $\{\tau_i\}_{i=1}^d$ are the absorbing coefficients. In practice, these directions can be
 113 the canonical basis of \mathbb{R}^d or potentially another set of varying orthonormal directions. In the former case,
 114 we refer to [35] for the discussion of the ‘‘Perfectly Matched’’ condition of this formulation. We also refer
 115 to [35] for the expression of the corresponding weak formulation. For solid domains, we consider the similar
 116 split first order formulation. In the scope of this paper, we cast aside the issue of PML stability, and we
 117 assume that the material properties \mathcal{C}^* in (5) allow for a stable use of this formulation. Readers may refer
 118 to [33, 34], and references therein, for more details on this matter.

119 2.2. Generalities on spectral finite elements

120 In this section we propose to go through the key notions of the finite element discretization. To simplify
 121 the presentation we consider mainly the scalar case and systems of the form of (2). This will enable us to
 122 introduce the principal notations in order to detail the benefits of the macro-element strategy later on.

123 2.2.1. Fully discrete scheme

124 We consider a Galerkin approximation space $V_h \subset V$ generated by the basis functions $\{\varphi_I\}_{I=1}^{N_h}$, with
 125 $N_h = \dim(V_h)$. We define the matrix form of the discrete (in space) counter part of (2) as

$$126 \quad \mathbb{M} \frac{d^2 \vec{P}}{dt^2} + \mathbb{K} \vec{P} = 0. \quad (8)$$

127 In (8), $\vec{P} \in \mathbb{R}^{N_h}$ is the vector regrouping the coefficients of the discrete solution in the basis of V_h . The
 128 matrices \mathbb{M} and \mathbb{K} are the so-called mass and stiffness matrices defined as

$$129 \quad \mathbb{M}_{IJ} = m(\varphi_I, \varphi_J), \quad \mathbb{K}_{IJ} = k(\varphi_I, \varphi_J), \quad \forall I, J = 1, \dots, N_h.$$

130 The fully discrete scheme is obtained by employing an explicit second-order leapfrog time scheme,

$$131 \quad \mathbb{M} \frac{\vec{P}^{n+1} - 2\vec{P}^n + \vec{P}^{n-1}}{\Delta t^2} + \mathbb{K} \vec{P}^n = 0, \quad (9)$$

132 which is stable upon the following Courant-Friedrichs-Lewy (CFL) condition, see e.g. [36] or Appendix A,

$$133 \quad \Delta t \leq \frac{2}{\sqrt{r(\mathbb{M}^{-1} \mathbb{K})}}, \quad (10)$$

134 where $r(\cdot)$ represents the spectral radius of a matrix. Note that this discretization procedure trivially extends
 135 to the elastodynamic case with relevant changes in the bilinear forms defined in (6). For PML formulations,
 136 we also use a time scheme consistent of order two, such as in [35].

137 *2.2.2. Spectral finite elements and mass lumping*

138 Lagrange finite element methods [31] propose a specific construction of the approximation space V_h based
 139 upon a given mesh \mathcal{T}_h of the computational domain. In the following, we consider meshes of hexahedral (or
 140 quadrilateral) elements satisfying the standard conformity constraints. The space V_h is defined as the space
 141 of functions that are globally continuous, and that have a local, i.e. per element, polynomial representative
 142 in a reference element. More precisely, assuming that each element $K \in \mathcal{T}_h$ is the result of a diffeomorphism
 143 \underline{F}_K applied onto a reference cube (or square) \widehat{K} , we have, for any $v_h \in V_h$,

$$144 \quad \forall K \in \mathcal{T}_h, \quad \exists \underline{k} = \underline{k}(K) \in \mathbb{N}^{*d}, \quad \exists! \widehat{v}_K \in \mathcal{Q}^{\underline{k}}(\widehat{K}), \quad v_h|_K = \widehat{v}_K \circ \underline{F}_K^{-1}. \quad (11)$$

145 In (11), $\mathcal{Q}^{\underline{k}}$ is the space of polynomials expressed as the tensor product of one-dimensional polynomial
 146 spaces, and we denote by \widehat{n}_h its dimension,

$$147 \quad \forall \underline{k} = (k_1 \cdots k_d)^\top \in \mathbb{N}^{*d}, \quad \mathcal{Q}^{\underline{k}}(\widehat{K}) = \prod_{p=1}^d \mathcal{P}^{k_p}([0; 1]), \quad \widehat{n}_h = \dim(\mathcal{Q}^{\underline{k}}) = \prod_{p=1}^d (k_p + 1).$$

148 We define on the reference element a set of \widehat{n}_h nodes $\widehat{\Xi} = \{\widehat{\xi}_i\}_{i=1}^{\widehat{n}_h}$, and we denote by $\{\widehat{\varphi}_i\}_{i=1}^{\widehat{n}_h}$ the associated
 149 local Lagrange polynomial basis. From the transformation of these local nodes by every \underline{F}_K , we obtain the
 150 set of global nodes $\Xi = \{\xi_I\}_{I=1}^{N_h}$, from which we have discarded every redundant coordinates at element
 151 boundaries. The global Lagrange basis functions of V_h are linked to the local Lagrange polynomials through

$$152 \quad \forall K \in \mathcal{T}_h \text{ s.t. } \xi_I \in K, \quad \exists! i_K \in \{1, \dots, \widehat{n}_h\}, \quad \varphi_I|_K = \widehat{\varphi}_{i_K} \circ \underline{F}_K^{-1}. \quad (12)$$

154 In (12), we have implicitly introduced the “local-to-global” index mapping

$$155 \quad \ell_G : \mathcal{T}_h \times \llbracket 1; \widehat{n}_h \rrbracket \longrightarrow \llbracket 1; N_h \rrbracket, \quad \text{s.t. } I = \ell_G(K, i_K). \quad (13)$$

156 Note that in (11), we allow functions in V_h to be locally represented by polynomials with orders potentially
 157 different in every directions. This particularity will be referred to as “anisotropic” orders of approximation.
 158 We will see in Section 5 that anisotropic orders enables a local adaptation of the discretization, in order to
 159 efficiently take into account thin layers of materials.

160 In the context of hexahedral (or quadrilateral) elements, a convenient way to define a proper set of
 161 Lagrange nodes $\widehat{\Xi}$ is by tensor product of d one-dimensional point distributions. The spectral finite element
 162 method is based upon Gauss-Lobatto points for defining these one-dimensional point distributions. Due to
 163 their optimal convergence as the order of approximation increases [27, 37], they have received a significant
 164 amount of interest in numerous fields of application. In the context of transient wave propagation modeling,
 165 spectral elements are particularly interesting since they allow for consistent “mass lumping” at any order of
 166 approximation [19, 20, 32, 38]. Broadly, this approach aims at approximating the mass matrix in (9) by a
 167 diagonal one, in order to obtain a fully explicit scheme. This approximation of the mass matrix is performed
 168 using a specific quadrature formula for computing the integrals appearing in the local mass matrices. The
 169 main conditions to achieve consistent mass lumping are [20, 32, 39]:

- 170 (i) the quadrature points and the nodes must coincide to obtain diagonal local mass matrices,
- 171 (ii) for stability reasons, the quadrature weights need to be strictly positive,
- 172 (iii) the quadrature formula must be exact at least for polynomials of order $2\underline{k} - 2$, in order to be as
 173 consistent as the case of an exact integration.

174 For hexahedral (or quadrilateral) elements, these three specific conditions are only satisfied by the Gauss-
 175 Lobatto quadrature formula. Spectral finite elements and mass lumping extend to elastodynamics [19, 32, 40]
 176 and PML formulations [33, 35].

177 *2.2.3. Local stiffness operations*

178 In practice, the diagonal mass matrix is assembled and stored in an initializing step of the numerical
 179 scheme. Hence, the main computational load in (9) comes from the application, at each time step, of
 180 the stiffness matrix to a finite element vector. Traditionally, the application of the global bilinear form is
 181 decomposed into the application of element-wise forms. Upon the assumption of a constant mass density of
 182 the fluid, we have

$$183 \quad k(v_h, w_h) = \frac{1}{\rho} \sum_{K \in \mathcal{T}_h} \widehat{k}_K(v_h, w_h), \quad \forall v_h, w_h \in V_h.$$

184 After a change of variable and using the local polynomial representatives of v_h and w_h , as in (11), the local
 185 stiffness operator reads

$$186 \quad \widehat{k}_K(v_h, w_h) = \int_{\widehat{K}} (\underline{\nabla}_{\widehat{x}} F_K^{-\top} \underline{\nabla}_{\widehat{x}} \widehat{v}_K) \cdot (\underline{\nabla}_{\widehat{x}} F_K^{-\top} \underline{\nabla}_{\widehat{x}} \widehat{w}_K) J_K \, d\widehat{\Omega}, \quad (14)$$

187 where $J_K = |\det(\underline{\nabla}_{\widehat{x}} F_K)|$. For conciseness, we introduce the notation

$$188 \quad \underline{\underline{G}}_K(\widehat{x}) = \frac{\text{co}(\underline{\nabla}_{\widehat{x}} F_K)^\top \text{co}(\underline{\nabla}_{\widehat{x}} F_K)}{J_K} \Big|_{\widehat{x}}, \quad \forall K \in \mathcal{T}_h, \quad \forall \widehat{x} \in \widehat{K}, \quad (15)$$

189 where $\text{co}(\cdot)$ is the cofactor matrix. As for the mass matrix, we use a Gauss-Lobatto quadrature formula
 190 to approximate the integral in (14). We denote by $\{\omega_q\}_{q=1}^{\widehat{n}_q}$ and $\{\widehat{x}_q\}_{q=1}^{\widehat{n}_q} \subset \widehat{K}$ the quadrature weights and
 191 points. The local stiffness operators (14) are approximated by

$$192 \quad \widehat{k}_K(v_h, w_h) \approx \widehat{k}_K^Q(v_h, w_h) = \sum_{q=1}^{\widehat{n}_h} \omega_q (\underline{\nabla}_{\widehat{x}} \widehat{v}_K^\top \underline{\underline{G}}_K \underline{\nabla}_{\widehat{x}} \widehat{w}_K) \Big|_{\widehat{\xi}_q}. \quad (16)$$

193 Denoting by $\{\alpha_{K,i}\}_{i=1}^{\widehat{n}_h}$ the coefficients of the local functions \widehat{v}_K in the Lagrange polynomial basis

$$194 \quad \widehat{v}_K(\widehat{x}) = \sum_{i=1}^{\widehat{n}_h} \alpha_{K,i} \widehat{\varphi}_i(\widehat{x}), \quad \forall \widehat{x} \in \widehat{K},$$

195 the expression of the corresponding gradient at the quadrature points reads

$$196 \quad \underline{\nabla}_{\widehat{x}} \widehat{v}_K(\widehat{\xi}_q) = \sum_{i=1}^{\widehat{n}_h} \alpha_{K,i} \underline{\nabla}_{\widehat{x}} \widehat{\varphi}_i(\widehat{\xi}_q).$$

197 Introducing the matrix $\widehat{\mathcal{D}} \in \mathcal{M}_{d \times n_h, n_h}(\mathbb{R})$ such that

$$198 \quad \forall i, j = 1, \dots, \widehat{n}_h, \quad \underline{\nabla}_{\widehat{x}} \widehat{\varphi}_i(\widehat{\xi}_j) = \left(\widehat{\mathcal{D}}_{d(i-1)+1,j} \cdots \widehat{\mathcal{D}}_{d(i-1)+d,j} \right)^\top \in \mathbb{R}^d, \quad (17)$$

199 the local finite element vectors representing \widehat{v}_K and its gradient are given by

$$200 \quad \overrightarrow{V}_K = (\alpha_{K,1} \cdots \alpha_{K,\widehat{n}_h})^\top, \quad \overrightarrow{\nabla V}_K = (\underline{\nabla}_{\widehat{x}} \widehat{v}_K(\widehat{\xi}_1) \cdots \underline{\nabla}_{\widehat{x}} \widehat{v}_K(\widehat{\xi}_{\widehat{n}_h}))^\top = \mathcal{D} \widehat{\varphi}^\top \overrightarrow{V}_K.$$

201 Regrouping in a matrix $\mathcal{G}_K \in \mathcal{M}_{d \times n_h, d \times n_h}(\mathbb{R})$ the evaluation of (15) at every local nodes times the quadrature
 202 weights, the local stiffness matrix (16) reads

$$203 \quad \widehat{k}_K^Q(v_h, w_h) = \overrightarrow{V}_K^\top \left(\widehat{\mathcal{D}} \mathcal{G}_K \widehat{\mathcal{D}}^\top \right) \overrightarrow{W}_K.$$

204 This decomposition extends to the elastic stiffness operator (6) and, for the sake of conciseness, is left to the
 205 reader. Similar arguments can also be used in the context of PML formulations in order to derive factorized
 206 forms of the bilinear operators appearing in the weak form of (7).

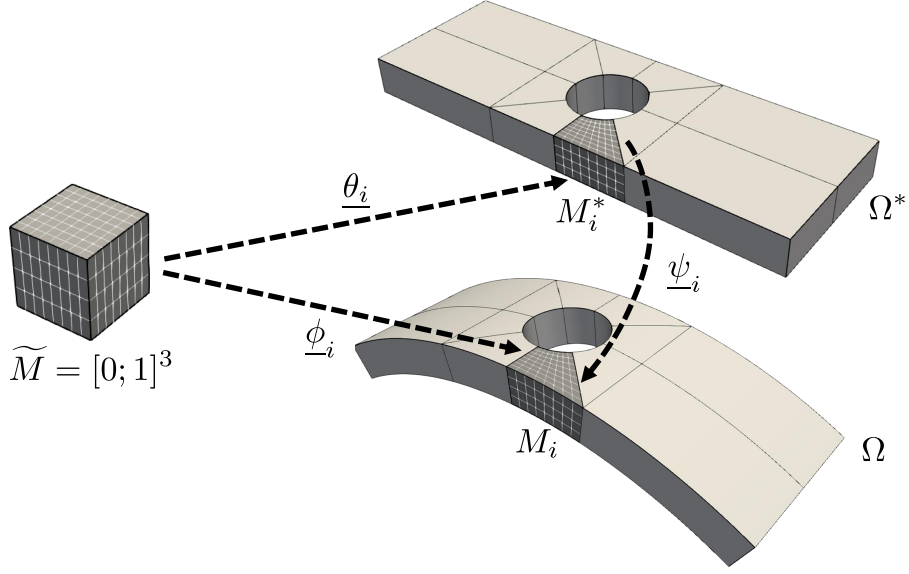


Figure 1: Illustration of a macro-element decomposition of a domain obtained from a cylindrical deformation of simple plate with a centered through-hole. The black lines in the geometry represent the edges of the macro-elements. The white lines represent the internal sub-discretization of a specific macro-element prior to and after deformation.

207 3. A macro-element strategy

208 3.1. Primary division of the computational domain

209 In the following, we consider a more complex configuration where, within the bounded domain Ω , mul-
 210 tiple formulations (fluid, solid or PML) can be considered. We assume that the computational domain is
 211 decomposed into N sub-domains,

$$212 \quad \Omega = \bigcup_{i=1}^N M_i,$$

213 and that each sub-domain M_i , referred to as a Macro-Element (ME), is built from a polynomial transfor-
 214 mation $\underline{\phi}_i$ of a reference cube (or square) \widetilde{M} ,

$$215 \quad \forall i = 1, \dots, N, \quad \exists \underline{k} = (k_1, \dots, k_d) \in \mathbb{N}^d, \quad \underline{\phi}_i \in [\mathcal{Q}^k(\widetilde{M})]^d.$$

216 This division of Ω is driven by the following constraints:

- 217 (i) The macro-mesh is a conform subdivision of Ω ,
- 218 (ii) To each interface between two different formulations corresponds an interface, or a set of interfaces,
 219 between two MEs or more,
- 220 (iii) The number of macro-elements should be minimal.

221 In Figure 1 we illustrate a macro-element decomposition of a curved plate having a through-hole at its
 222 center. In this example, the macro-mesh is made of twelve MEs, deformed from the unit cube in order to
 223 fit the geometry of the hole.

224 In some cases, we may consider an additional set of polynomial transformations $\underline{\theta}_i$ of the reference macro-
 225 element. These transformations can be used to represent the computational domain prior to some global
 226 deformation $\underline{\Psi}$. Let us denote by Ω^* the domain in this prior state such that

$$227 \quad \Omega = \underline{\Psi}(\Omega^*), \quad \Omega^* = \bigcup_{i=1}^N M_i^* = \bigcup_{i=1}^N \underline{\theta}_i(\widetilde{M}).$$

228 In Figure 1, we illustrate the case where Ω^* represents the plane geometry prior to a simple cylindrical
 229 deformation. In concrete applications, we might be interested in global deformations $\underline{\Psi}$ that can be a
 230 combination of cylindrical deformations, a spline deformation, or potentially more complex alterations of
 231 Ω^* . The effect of this deformation onto a macro-element M_i^* can be represented by the pair of polynomial
 232 transformations $\underline{\theta}_i$ and $\underline{\phi}_i$, i.e.

$$\underline{\psi}_i = \underline{\Psi}|_{M_i^*} = \underline{\phi}_i \circ \underline{\theta}_i^{-1}.$$

234 3.2. Sub-division of the macro-elements

235 Due to its “minimal” nature, the division of the computational domain proposed previously is unfit for
 236 adequate numerical procedures at the wavelength scale. Therefore, we resort to an internal sub-division of
 237 each macro-element M_i . It corresponds to a grid of hexahedral (or quadrilateral) cells deformed by $\underline{\phi}_i$. Let
 238 us denote by $\tilde{\mathcal{T}}_{h,i}$ and $\mathcal{T}_{h,i}$ the grid prior to and after its deformation so that, formally,

$$\mathcal{T}_{h,i} = \underline{\phi}_i(\tilde{\mathcal{T}}_{h,i}), \quad \forall i = 1, \dots, N. \quad (18)$$

240 Each grid cell $\tilde{K} \in \tilde{\mathcal{T}}_{h,i}$ is obtained from the reference element by a simple affine transformation $\underline{F}_{\tilde{K}}$. For
 241 instance, for $d = 3$, it reads

$$\underline{F}_{\tilde{K}}(\hat{\underline{x}}) = \begin{pmatrix} O_1(\tilde{K}) \\ O_2(\tilde{K}) \\ O_3(\tilde{K}) \end{pmatrix} + \begin{pmatrix} h_1(\tilde{K}) & & \\ & h_2(\tilde{K}) & \\ & & h_3(\tilde{K}) \end{pmatrix} \hat{\underline{x}}, \quad \forall \hat{\underline{x}} \in \hat{K}, \quad (19)$$

243 where $\{O_p(\tilde{K})\}_{p=1}^d$ and $\{h_p(\tilde{K})\}_{p=1}^d$ are the origin and the lengths of the cell respectively. With these
 244 notations, for each element $K \in \mathcal{T}_{h,i}$, the transformation \underline{F}_K introduced in Section 2.2.2 is expressed as

$$\underline{F}_K(\hat{\underline{x}}) = (\underline{\phi}_i \circ \underline{F}_{\tilde{K}})(\hat{\underline{x}}), \quad \forall \hat{\underline{x}} \in \hat{K}. \quad (20)$$

246 To each element in $\mathcal{T}_{h,i}$, we associate a spectral finite element with anisotropic orders. We denote by $N_{h,i}$
 247 the number of nodes and by

$$\Xi_i = \{\xi_I^i\}_{I=1}^{N_{h,i}}, \quad \forall i = 1, \dots, N,$$

249 the coordinates of the nodes. Note that (18) and (20) naturally extend to the polynomial transformation
 250 $\underline{\theta}_i$ of the macro-element so that we can define a mesh $\mathcal{T}_{h,i}^*$ of M_i^* , composed of the nodes Ξ_i^* . In practice,
 251 the refinement of the grid depends, for precision purposes, on an *a priori* estimation of the wavelengths of
 252 interest and on the maximal deformation of the grid. Note that, in (19), the lengths may vary from one
 253 cell to another, as long as $\tilde{\mathcal{T}}_{h,i}$ remains a conform grid. This will be of major importance when dealing with
 254 stratified materials in Section 5.

255 3.3. Assumption of conform interfaces

256 Throughout this paper, we make the assumption that the union of every macro-element sub-divisions,

$$\mathcal{T}_h = \bigcup_{i=1}^N \mathcal{T}_{h,i},$$

258 generates a conform mesh of the computational domain after discarding redundant coordinates at interfaces.
 259 In particular, let M_i and M_j be two macro-elements having a face in common, say Γ , then there exists a
 260 one-to-one mapping between the coordinates of the nodes $\Xi_i|_{\Gamma}$ and $\Xi_j|_{\Gamma}$. Let us denote by M_h the number
 261 of nodes on the interface Γ , we can introduce the “interface-to-volume” index mappings \mathcal{V}_{Γ}^i and \mathcal{V}_{Γ}^j such
 262 that

$$\forall k \in \{i, j\} \quad \mathcal{V}_{\Gamma}^k : \llbracket 1; M_h \rrbracket \longrightarrow \llbracket 1; N_{h,k} \rrbracket, \quad \text{and} \quad \forall I = 1, \dots, M_h, \quad \xi_{\mathcal{V}_{\Gamma}^i(I)}^i = \xi_{\mathcal{V}_{\Gamma}^j(I)}^j. \quad (21)$$

264 Alternatively, these mappings may be represented by the rectangular matrices \mathbb{V}_Γ^i and \mathbb{V}_Γ^j of dimension
 265 $N_{h,i} \times M_h$ and $N_{h,j} \times M_h$ respectively, such that

$$266 \quad \forall k \in \{i, j\} \quad \forall I = 1, \dots, N_{h,k}, \quad \forall J = 1, \dots, M_h \quad (\mathbb{V}_\Gamma^k)_{IJ} = \begin{cases} 1, & \text{if } \underline{\xi}_I^k \in \Gamma \text{ and } I = \mathcal{V}_\Gamma^k(J), \\ 0, & \text{otherwise.} \end{cases} \quad (22)$$

267 This aspect, referred to in the following as the conform interface assumption, will have important conse-
 268 quences when using the mortar element method in Section 4.

269 3.4. Benefits of the macro-element strategy

270 In order to cope with our objective (G2), we favor low-memory strategies. In this regard, we consider
 271 “unassembled” operations for representing the application of the stiffness matrix to an input finite element
 272 vector. In essence, it corresponds to applying every local stiffness matrices, described in Section 2.2.3, to
 273 local input vectors. In the simplified configuration where the sub-domains bear an acoustic formulation, we
 274 propose the pseudo-code in Algorithm 1 for performing these operations. In this pseudo-code, $\{\vec{W}^i\}_{i=1}^N$ rep-
 275 resent input finite element vectors, and $\{\vec{V}^i\}_{i=1}^N$ are the output vectors storing the results of the application
 276 of the stiffness matrices. We also consider in Algorithm 1 a standard coloring of each element in $\mathcal{T}_{h,i}$ so that
 no elements in a color group have nodes in common.

```

Input :  $\vec{W}^1, \dots, \vec{W}^N$ 
Output:  $\vec{V}^1, \dots, \vec{V}^N$ 
for  $i \leftarrow 1$  to  $N$  do
  foreach color group in  $\mathcal{T}_{h,i}$  do
    foreach element  $K$  in color group do
      for  $j \leftarrow 1$  to  $\hat{n}_h$  do  $\vec{W}_K^i[j] \leftarrow \vec{W}^i[\ell_G(K, j)]$ 
       $\vec{V}_K^i \leftarrow (\hat{\mathcal{D}} \mathcal{G}_K \hat{\mathcal{D}}^\top) \vec{W}_K^i$ 
      for  $j \leftarrow 1$  to  $\hat{n}_h$  do  $\vec{V}^i[\ell_G(K, j)] \leftarrow \vec{V}^i[\ell_G(K, j)] + \vec{V}_K^i[j]$ 
    end
  end
end

```

Algorithm 1: Pseudo-code for unassembled stiffness operations.

277 These unassembled operations are rather natural in the context of high-frequency wave propagation
 278 modeling. However, our approach enables some significant improvements, mainly due to the fact that the
 279 underlying sub-discretization of each macro-element is related to a reference grid in \tilde{M} . In particular,
 280 while in a general case of unstructured meshes, the “local-to-global” mapping ℓ_G defined in (13) is usually
 281 stored, it can be recomputed at no expense within each macro-element, thus sparing important memory
 282 load. Additionally, the matrix \mathcal{G}_K in (15), built from the transformation of the current element, can also
 283 be recomputed on-the-fly. To do so, we simply apply the transformation (20) to the local nodes coordinates
 284 $\hat{\Xi}$ and compute the gradient of the transformation using $\hat{\mathcal{D}}^\top$, defined in (17). Thus, most of the memory
 285 footprint of a macro-element comes from the three finite element vectors of the time scheme and the diagonal
 286 mass matrix. Note that in this context, the major part of the workload in Algorithm 1 is centered on
 287 computing gradients of local finite element vectors. In practice we use a specific data structure in order to
 288 improve this operation, readers may refer to [41] for more details. On top of improving the memory load
 289 one can also increase the CPU performance by applying the local stiffness operations in parallel. Indeed,
 290 due to the definition of the elements coloring, the third loop in Algorithm 1 is naturally fit for parallelism.
 291 Since the topology of the mesh of a single ME is essentially a grid, this coloring is trivial and optimal in
 292 the sense that: (1) each color group has the same number of elements; (2) the number of different color is
 293 minimal.
 294

Another interesting aspect of the macro-element strategy resides in its capacity to reconstruct efficiently and on-the-fly specific local vector fields. An important example is when a constant unitary vector, say \underline{v}^* , is defined on a macro-element M_i^* prior to its deformation by $\underline{\Psi}$. In this context, we can retrieve its local variation within the final macro-element through

$$\underline{v}|_{M_i} = \frac{(\underline{\nabla}_{\underline{x}} \phi_i \underline{\nabla}_{\underline{x}^*} \varrho_i^{-1}) \underline{v}^*}{\|(\underline{\nabla}_{\underline{x}} \phi_i \underline{\nabla}_{\underline{x}^*} \varrho_i^{-1}) \underline{v}^*\|}, \quad \forall i = 1, \dots, N. \quad (23)$$

A natural application is the case of fiber materials, where specific material properties are defined in a privileged direction, the fiber, while being isotropic in the orthogonal plane. One may typically define the constant fiber orientation in Ω^* and compute the result of its transformation from (23). From the local fiber orientation we can reconstruct any two orthogonal directions and define a local isotropic transverse constitutive law using (5). In Section 5, we present numerical results using this computational strategy to model curved stratified composites. A similar configuration is when a constant unitary vector, say $\underline{\tilde{v}}$, is defined in the reference macro-element \widetilde{M} . We can retrieve, within each macro element, its local deformation from

$$\underline{v}|_{M_i} = \frac{\underline{\nabla}_{\underline{x}} \phi_i \underline{\tilde{v}}}{\|\underline{\nabla}_{\underline{x}} \phi_i \underline{\tilde{v}}\|}, \quad \forall i = 1, \dots, N. \quad (24)$$

This can typically be used to compute splitting directions of PML systems of the form of (7). Operating in this fashion, we encompass in the same framework curved and straight PML domains with an automatic and lightweight definition of the local directions.

Low memory footprint, parallel operations on standard CPU and automatic reconstruction of relevant modeling components form the main benefits of our approach that enable us to address our primary goals (G1) and (G2). These advantages can be significant especially for 3D configurations, but they require a “minimal” high-order hexahedral (or quadrilateral) decomposition of Ω , which is known to be a major challenge in the field of computational geometry. However, since our goal is to achieve parametric studies, we do not intend to provide efficient numerical procedures for fully generic configurations. In most of the practical cases we restrict ourselves to parametric geometries that can be obtained from case-dependent meshing procedures. Readers may refer to [42, 43, 44, 45] for some examples in the context of UT modeling.

4. Coupling macro-element formulations using mortar elements

One of the challenge of our approach is to enable the communication between neighboring sub-domains. Upon the conform interface assumption stated in Section 3.3, the difficulty essentially lies in coupling different formulations within the same framework. To address this issue we rely on the mortar element method [46, 47, 48, 49]. This domain decomposition method has been successfully used as a mean of incorporating different and independent space discretizations within a global numerical scheme. A significant amount of research, see for instance [28, 29, 30, 50, 51], has been dedicated to the analysis and application of mortar elements for elastodynamics and wave propagation problems. In our work, we use this method for our specific objectives, and we show how one can minimize its computational cost by employing a lumping technique, as the one described in Section 2.2.2.

4.1. Schur complement

To start with, we recall the Schur complement method, which will be used to devise a global algorithm for coupling macro-element formulations. We consider two vector of unknowns: $\vec{X} \in \mathbb{R}^N$, referred to as the volume unknown, and $\vec{L} \in \mathbb{R}^M$, referred to as the interface unknown. We assume that they satisfy the following linear system

$$\begin{pmatrix} \mathbb{D} & \mathbb{B} \\ \mathbb{C}^\top & \mathbb{D}^\epsilon \end{pmatrix} \begin{pmatrix} \vec{X} \\ \vec{L} \end{pmatrix} = \begin{pmatrix} \vec{F} \\ \vec{G} \end{pmatrix}, \quad (25)$$

336 where \vec{F} and \vec{G} are given right-hand sides. The matrices \mathbb{D} and \mathbb{D}^ε , of dimension $N \times N$ and $M \times M$
 337 respectively, are diagonal. The matrices \mathbb{B} and \mathbb{C} of dimension $N \times M$ may be referred to as transmission
 338 (from interface to volume) matrices. The Schur complement method for solving this type of linear system
 339 reads:

340 **Step 1. Pre-processing.** We compute an auxiliary variable $\vec{X}^* \in \mathbb{R}^N$ by (trivial) inversion of the volume
 341 matrix

$$342 \quad \vec{X}^* = \mathbb{D}^{-1} \vec{F}. \quad (26)$$

343 **Step 2. Computation.** Defining the Schur complement matrix

$$344 \quad \mathbb{S}^\varepsilon = \mathbb{D}^\varepsilon - \mathbb{C}^\top \mathbb{D}^{-1} \mathbb{B}, \quad (27)$$

345 and assuming that this matrix is invertible, we solve the following linear system

$$346 \quad \vec{L} = (\mathbb{S}^\varepsilon)^{-1} \left\{ \vec{G} - \mathbb{C}^\top \vec{X}^* \right\}. \quad (28)$$

347 **Step 3. Post-processing.** The volume unknown is obtained by

$$348 \quad \vec{X} = \vec{X}^* - \mathbb{D}^{-1} \mathbb{B} \vec{L}.$$

349 Note that, in these three steps, the main issues are the invertibility of the Schur complement matrix \mathbb{S}^ε in
 350 (27), and, if it is invertible, the efficient computation of the interface unknown solution of (28).

351 4.2. Two-domain problems

352 To start with, we consider the case of two sub-domains, i.e. $\Omega = M_1 \cup M_2$, connected at an interface
 353 Γ satisfying the conform interface assumption, detailed in Section 3.3. We denote by $\underline{\nu}$ the normal vector
 354 field of the interface oriented from M_1 to M_2 .

355 4.2.1. The illustrative example of fluid-fluid coupling

356 We assume that both M_1 and M_2 support a fluid formulation (1), satisfied by the corresponding pressure
 357 fields p_1 and p_2 with mass densities ρ_1 and ρ_2 and sound velocities c_1 and c_2 . In this case, the mortar element
 358 method introduces a Lagrange multiplier $\lambda \in W = H^{-\frac{1}{2}}(\Gamma)$, satisfying [52]

$$359 \quad \frac{\partial \lambda}{\partial t} = \frac{1}{\rho_1} \nabla p_1 \cdot \underline{\nu} = \frac{1}{\rho_2} \nabla p_2 \cdot \underline{\nu}, \quad (29)$$

360 and weakly imposes the continuity condition on p_1 and p_2 at the interface. We define, for $i = 1, 2$, the space
 361 $V_i = H^1(M_i)$, and the product space $X = V_1 \times V_2 \times W$. The weak formulation of the coupling scheme aims
 362 at finding the solution $(p_1, p_2, \lambda) \in X$ for a time $t > 0$ such that, for any $(p_1^*, p_2^*, \lambda^*) \in X$, we have

$$363 \quad \begin{cases} \frac{d^2}{dt^2} m_1(p_1, p_1^*) + k_1(p_1, p_1^*) - \frac{d}{dt} \int_\Gamma \lambda p_1^* d\Gamma = 0, \\ \frac{d^2}{dt^2} m_2(p_2, p_2^*) + k_2(p_2, p_2^*) + \frac{d}{dt} \int_\Gamma \lambda p_2^* d\Gamma = 0, \\ \frac{d}{dt} \int_\Gamma (p_1 - p_2) \lambda^* d\Gamma + \varepsilon \frac{d^2}{dt^2} \int_\Gamma \lambda \lambda^* d\Gamma = 0. \end{cases} \quad (30)$$

364 In (30), we have assumed free-surface boundary conditions on the remaining faces of the macro-elements.
 365 Also, $m_i(\cdot, \cdot)$ and $k_i(\cdot, \cdot)$ are the bilinear forms defined as in (3), with corresponding material properties ρ_i
 366 and c_i . Note that, in this formulation, we have introduced in the weak continuity constraint a penalization
 367 term proportional to a (small) positive constant coefficient $\varepsilon \geq 0$ and the second time-derivative of the
 368 Lagrange multiplier. This specific choice of the penalization term is justified in Appendix A, where we
 369 consider the fully discrete energy norm of the coupling system.

370 We define $V_{h,i} \subset V_i$ the finite element approximation spaces, detailed in Section 2.2.2, and we denote
 371 by $\{\varphi_{i,I}\}_{I=1}^{N_{h,i}}$ the associated basis functions. The approximation space of W is defined by taking the trace
 372 space of $V_{h,1}$ onto the interface, namely

$$373 \quad W_h = V_{h,1}|_\Gamma, \quad W_h = \text{span} \{\mu_I\}_{I=1}^{M_h}, \quad M_h = \dim(W_h). \quad (31)$$

374 As a consequence there is a trivial mapping between the basis function of W_h and the basis functions of
 375 $V_{h,1}$. Upon the conform interface assumption, this trivial mapping extends to basis functions of $V_{h,2}$, and,
 376 using the “interface-to-volume” mapping (21), we have

$$377 \quad \mu_I = \varphi_{1,\mathcal{V}_1^1(I)}|_\Gamma = \varphi_{2,\mathcal{V}_2^1(I)}|_\Gamma, \quad \forall I = 1, \dots, M_h.$$

378 In the context of the mortar element method, our choice of discrete space for the Lagrange multipliers is
 379 non-standard and usually prohibited, since it may lead, in general configurations, to an unsatisfied discrete
 380 *inf-sup* condition [53, 54]. It is for this particular reason that the penalization term was introduced in (30),
 381 in order to retrieve the stability of the discrete coupling scheme.

382 Applying a second order time-scheme, such as the one presented in (9), and a centered scheme for the
 383 coupling terms lead to the following fully-discrete coupling scheme

$$384 \quad \begin{cases} \frac{1}{\Delta t^2} \mathbb{M}_1 \{ \vec{P}_1^{n+1} - 2\vec{P}_1^n + \vec{P}_1^{n-1} \} + \mathbb{K}_1 \vec{P}_1^n - \frac{1}{2\Delta t} \mathbb{C}_1 \{ \vec{\Lambda}^{n+1} - \vec{\Lambda}^{n-1} \} = 0, \\ \frac{1}{\Delta t^2} \mathbb{M}_2 \{ \vec{P}_2^{n+1} - 2\vec{P}_2^n + \vec{P}_2^{n-1} \} + \mathbb{K}_2 \vec{P}_2^n + \frac{1}{2\Delta t} \mathbb{C}_2 \{ \vec{\Lambda}^{n+1} - \vec{\Lambda}^{n-1} \} = 0, \\ \frac{1}{2\Delta t} \mathbb{C}_1 \{ \vec{P}_1^{n+1} - \vec{P}_1^{n-1} \} - \frac{1}{2\Delta t} \mathbb{C}_2 \{ \vec{P}_2^{n+1} - \vec{P}_2^{n-1} \} + \frac{\varepsilon}{\Delta t^2} \mathbb{M}_\Gamma \{ \vec{\Lambda}^{n+1} - 2\vec{\Lambda}^n + \vec{\Lambda}^{n-1} \} = 0. \end{cases} \quad (32)$$

385 In (32), \mathbb{M}_i and \mathbb{K}_i are the mass and stiffness matrices associated to each formulation, $\{\vec{P}_i^k\}_{k \geq 0}$ are the
 386 finite element unknowns expressed on each volume, and $\{\vec{\Lambda}^k\}_{k \geq 0}$ are the discrete Lagrange multipliers. The
 387 interface mass matrix $\mathbb{M}_\Gamma \in \mathcal{M}_{M_h, M_h}(\mathbb{R})$ is defined by

$$388 \quad (\mathbb{M}_\Gamma)_{IJ} = \int_\Gamma \mu_I \mu_J \, d\Gamma, \quad \forall I, J = 1, \dots, M_h. \quad (33)$$

389 From the choice of the discrete space for the Lagrange multipliers, we have

$$390 \quad (\mathbb{M}_\Gamma)_{IJ} = \int_\Gamma (\varphi_{1,\mathcal{V}_1^1(I)}|_\Gamma) (\varphi_{1,\mathcal{V}_1^1(J)}|_\Gamma) \, d\Gamma,$$

391 where $\mathcal{V}_\Gamma^1(\cdot)$ is the “interface-to-volume” index mapping defined in (21). Hence, applying the same consistent
 392 mass lumping technique as for the volume mass matrix, presented in Section 2.2.2, we obtain a diagonal
 393 interface mass matrix. The transmission matrices $\mathbb{C}_i \in \mathcal{M}_{N_{h,i}, M_h}(\mathbb{R})$ are expressed as

$$394 \quad (\mathbb{C}_i)_{IJ} = \int_\Gamma (\varphi_{i,I}|_\Gamma) \mu_J \, d\Gamma, \quad \forall I = 1, \dots, N_{h,i}, \quad \forall J = 1, \dots, M_h, \quad (34)$$

395 and, using the matrix representation of the “interface-to-volume” index mappings (22), can be expressed as

$$396 \quad \mathbb{C}_i = \mathbb{V}_\Gamma^i \mathbb{M}_\Gamma. \quad (35)$$

398 **Remark 1.** In Appendix A, we detail the arguments proving that, in order for the discrete system (32) to
 399 be stable, the timestep must respect the CFL condition

$$400 \quad \Delta t \leq \min \left\{ \frac{2}{\sqrt{r(\mathbb{M}_1^{-1} \mathbb{K}_1)}}, \frac{2}{\sqrt{r(\mathbb{M}_2^{-1} \mathbb{K}_2)}} \right\}. \quad (36)$$

401 In other words, the introduction of the Lagrange multipliers does not modify the stability condition on the
 402 timestep deriving from both sub-domains, which is a major asset of this domain decomposition method.

403 The discrete scheme (32) can be expressed as

$$404 \begin{pmatrix} \mathbb{M}_1 & & -\frac{\Delta t}{2}\mathbb{C}_1 \\ & \mathbb{M}_2 & \frac{\Delta t}{2}\mathbb{C}_2 \\ \frac{\Delta t}{2}\mathbb{C}_1^\top & -\frac{\Delta t}{2}\mathbb{C}_2^\top & \varepsilon\mathbb{M}_\Gamma \end{pmatrix} \begin{pmatrix} \vec{P}_1^{n+1} \\ \vec{P}_2^{n+1} \\ \vec{\Lambda}^{n+1} \end{pmatrix} = \begin{pmatrix} \vec{F}_1^{n,n-1} \\ \vec{F}_2^{n,n-1} \\ \vec{G}^{n,n-1} \end{pmatrix}, \quad (37)$$

405 where the right-hand sides are

$$406 \begin{cases} \vec{F}_i^{n,n-1} = -\Delta t^2 \mathbb{K}_i \vec{P}_i^n + \mathbb{M}_i \{2\vec{P}_i^n - \vec{P}_i^{n-1}\} \pm \frac{\Delta t}{2} \mathbb{C}_i \vec{\Lambda}^{n-1}, & \forall i = 1, 2, \\ \vec{G}^{n,n-1} = \frac{\Delta t}{2} \{ \mathbb{C}_1^\top \vec{P}_1^{n-1} - \mathbb{C}_2^\top \vec{P}_2^{n-1} \} + \varepsilon \{2\vec{\Lambda}^n - \vec{\Lambda}^{n-1}\}. \end{cases} \quad (38)$$

407 System (37) is a specific case of system (25) with unknowns

$$408 \vec{X} = \begin{pmatrix} \vec{P}_1^{n+1} \\ \vec{P}_2^{n+1} \end{pmatrix}, \quad \vec{L} = \vec{\Lambda}^{n+1},$$

409 and matrices

$$410 \mathbb{D} = \begin{pmatrix} \mathbb{M}_1 & \\ & \mathbb{M}_2 \end{pmatrix}, \quad \mathbb{D}^\varepsilon = \varepsilon\mathbb{M}_\Gamma, \quad \mathbb{B} = \frac{\Delta t}{2} \begin{pmatrix} -\mathbb{C}_1 \\ \mathbb{C}_2 \end{pmatrix}, \quad \mathbb{C} = -\mathbb{B}.$$

411 Hence, the Schur complement matrix reads

$$412 \mathbb{S}^\varepsilon = \varepsilon\mathbb{M}_\Gamma + \frac{\Delta t^2}{4} \sum_{i=1,2} \mathbb{C}_i^\top \mathbb{M}_i^{-1} \mathbb{C}_i. \quad (39)$$

413 Since the interface mass matrix and the volume mass matrices are diagonal, using the decomposition (35)
414 of the transmission matrices, and remarking that $(\mathbb{V}_\Gamma^i)^\top (\mathbb{V}_\Gamma^i)$ is the identity matrix, one can verify that the
415 Schur complement matrix is diagonal,

$$416 (\mathbb{S}^\varepsilon)_{II} = \varepsilon(\mathbb{M}_\Gamma)_{II} + \frac{\Delta t^2}{4} \sum_{i=1,2} \frac{(\mathbb{M}_\Gamma)_{II}^2}{(\mathbb{M}_i)_{\mathbb{V}_\Gamma^i(I)\mathbb{V}_\Gamma^i(I)}}, \quad \forall I = 1, \dots, M_h. \quad (40)$$

417 This lumped Schur complement matrix is invertible for $\varepsilon \geq 0$, and its memory footprint is very limited.
418 Hence, the coupling procedures represent little computational overheads.

419 4.2.2. The case of fluid-solid coupling

420 We now consider the case where M_1 supports a fluid formulation (1), and M_2 a solid formulation (4). We
421 introduce two Lagrange multipliers λ_v and λ_s , used to solve the normal velocity constraint and the normal
422 stress constraint respectively,

$$423 \lambda_v = \underline{v}_1 \cdot \underline{\nu} = \frac{\partial y_2}{\partial t} \cdot \underline{\nu}, \quad \lambda_s \underline{\nu} = -p_1 \underline{\nu} = \underline{\sigma}_2 \cdot \underline{\nu},$$

424 where \underline{v}_1 is the velocity associated to the fluid acoustic pressure in M_1 . The first relation can be written in
425 terms of the pressure unknown since

$$426 \frac{\partial \lambda_v}{\partial t} = \frac{\partial \underline{v}_1}{\partial t} \cdot \underline{\nu} = -\frac{1}{\rho_1} \nabla p_1 \cdot \underline{\nu}.$$

427 Defining $V_1 = H^1(M_1)$, $V_2 = [H^1(M_2)]^d$, $W = H^{-\frac{1}{2}}(\Gamma)$ and the product space $X = V_1 \times V_2 \times W \times W$, the weak
428 formulation of the coupling scheme, satisfied by the solution $(p_1, \underline{y}_2, \lambda_v, \lambda_s) \in X$ for any $(p_1^*, \underline{y}_2^*, \lambda_v^*, \lambda_s^*) \in X$,

429 reads

$$\begin{cases}
\frac{d^2}{dt^2} m_1(p_1, p_1^*) + k_1(p_1, p_1^*) + \frac{d}{dt} \int_{\Gamma} \lambda_v p_1^* d\Gamma = 0, \\
\frac{d^2}{dt^2} m_2(\underline{y}_2, \underline{y}_2^*) + k_2(\underline{y}_2, \underline{y}_2^*) + \int_{\Gamma} \lambda_s (\underline{y}_2^* \cdot \underline{\nu}) d\Gamma = 0, \\
\int_{\Gamma} (\lambda_v - \frac{\partial \underline{y}_2}{\partial t} \cdot \underline{\nu}) \lambda_v^* d\Gamma = 0, \\
\int_{\Gamma} (\lambda_s + p_1) \lambda_s^* d\Gamma = 0.
\end{cases} \quad (41)$$

431 In (41), $m_2(\cdot, \cdot)$ and $k_2(\cdot, \cdot)$ are the bilinear forms defined in (6). Compared to (30), no penalization term is
432 needed since the normal velocity and normal stress weak continuity relations directly involve coercive terms
433 w.r.t. the Lagrange multipliers.

434 We define the discrete space W_h as in (31). Using an order two centered time scheme for the comple-
435 mentary terms appearing in (41), we obtain the following fully discrete scheme

$$\begin{cases}
\frac{1}{\Delta t^2} \mathbb{M}_1 \{ \vec{P}_1^{n+1} - 2\vec{P}_1^n + \vec{P}_1^{n-1} \} + \mathbb{K}_1 \vec{P}_1^n + \frac{1}{2\Delta t} \mathbb{C}_1 \{ \vec{\Lambda}_v^{n+1} - \vec{\Lambda}_v^{n-1} \} = 0, \\
\frac{1}{\Delta t^2} \mathbb{M}_2 \{ \vec{Y}_2^{n+1} - 2\vec{Y}_2^n + \vec{Y}_2^{n-1} \} + \mathbb{K}_2 \vec{Y}_2^n + \frac{1}{2} \mathbb{C}_{2,\underline{\nu}} \{ \vec{\Lambda}_s^{n+1} + \vec{\Lambda}_s^n \} = 0, \\
\frac{1}{2} \mathbb{M}_{\Gamma} \{ \vec{\Lambda}_v^{n+1} + \vec{\Lambda}_v^n \} - \frac{1}{2\Delta t} \mathbb{C}_{2,\underline{\nu}}^{\top} \{ \vec{Y}_2^{n+1} - \vec{Y}_2^{n-1} \} = 0, \\
\frac{1}{2} \mathbb{M}_{\Gamma} \{ \vec{\Lambda}_s^{n+1} + \vec{\Lambda}_s^n \} + \frac{1}{2} \mathbb{C}_1^{\top} \{ \vec{P}_1^{n+1} + \vec{P}_1^n \} = 0.
\end{cases} \quad (42)$$

437 In (42), the matrices \mathbb{M}_{Γ} and \mathbb{C}_1 are defined in (33) and (34) and respectively. The matrix $\mathbb{C}_{2,\underline{\nu}} \in$
438 $\mathcal{M}_{N_{h,2}, M_h}(\mathbb{R})$ is defined as

$$(\mathbb{C}_{2,\underline{\nu}})_{IJ} = \int_{\Gamma} (\varphi_{2,I} \cdot \underline{\nu}) \mu_J d\Gamma, \quad \forall I = 1, \dots, N_{h,2}, \quad \forall J = 1, \dots, M_h,$$

440 where $\{\varphi_{2,I}\}_{I=1}^{N_{h,2}}$ are the (vectorial) basis functions generating $V_{h,2}$. Using similar energy arguments as in
441 Appendix A, one can prove that the CFL condition on the timestep, for the discrete system (42) to be
442 stable, is the lowest CFL condition of the two sub-domains computed independently. The numerical scheme
443 (42) is equivalent to

$$\begin{pmatrix} \mathbb{M}_1 & & \frac{\Delta t}{2} \mathbb{C}_1 & & \\ & \mathbb{M}_2 & & \frac{\Delta t^2}{2} \mathbb{C}_{2,\underline{\nu}} & \\ & -\frac{1}{\Delta t} \mathbb{C}_{2,\underline{\nu}}^{\top} & \mathbb{M}_{\Gamma} & & \\ \mathbb{C}_1^{\top} & & & \mathbb{M}_{\Gamma} & \end{pmatrix} \begin{pmatrix} \vec{P}_1^{n+1} \\ \vec{Y}_2^{n+1} \\ \vec{\Lambda}_v^{n+1} \\ \vec{\Lambda}_s^{n+1} \end{pmatrix} = \begin{pmatrix} \vec{F}_1^{n,n-1} \\ \vec{F}_2^{n,n-1} \\ \vec{G}_v^{n,n-1} \\ \vec{G}_s^n \end{pmatrix}, \quad (43)$$

445 where the right-hand sides related to the volume unknowns are expressed as

$$\begin{cases} \vec{F}_1^{n,n-1} = -\Delta t^2 \mathbb{K}_1 \vec{P}_1^n + \mathbb{M}_1 \{ 2\vec{P}_1^n - \vec{P}_1^{n-1} \} + \frac{\Delta t}{2} \mathbb{C}_1 \vec{\Lambda}_v^{n-1}, \\ \vec{F}_2^{n,n-1} = -\Delta t^2 \mathbb{K}_2 \vec{Y}_2^n + \mathbb{M}_2 \{ 2\vec{Y}_2^n - \vec{Y}_2^{n-1} \} - \frac{\Delta t^2}{2} \mathbb{C}_{2,\underline{\nu}} \vec{\Lambda}_s^n, \end{cases}$$

447 while the right-hand sides associated to the discrete Lagrange multipliers are

$$\begin{cases} \vec{G}_v^{n,n-1} = -\frac{1}{\Delta t} \mathbb{C}_{2,\underline{\nu}}^{\top} \vec{Y}_2^{n-1} - \mathbb{M}_{\Gamma} \vec{\Lambda}_v^n, \\ \vec{G}_s^n = -\mathbb{C}_1^{\top} \vec{P}_1^n - \mathbb{M}_{\Gamma} \vec{\Lambda}_s^n. \end{cases}$$

449 System (43) is a specific case of system (25) with unknowns

$$\vec{X} = \begin{pmatrix} \vec{P}_1^{n+1} \\ \vec{Y}_2^{n+1} \end{pmatrix}, \quad \vec{L} = \begin{pmatrix} \vec{\Lambda}_v^{n+1} \\ \vec{\Lambda}_s^{n+1} \end{pmatrix}.$$

451 The diagonal and extra-diagonal block matrices are

$$452 \quad \mathbb{D} = \begin{pmatrix} \mathbb{M}_1 & \\ & \mathbb{M}_2 \end{pmatrix}, \quad \mathbb{D}^\varepsilon = \begin{pmatrix} \mathbb{M}_\Gamma & \\ & \mathbb{M}_\Gamma \end{pmatrix}, \quad \mathbb{B} = \begin{pmatrix} \frac{\Delta t}{2} \mathbb{C}_1 & \\ & \frac{\Delta t^2}{2} \mathbb{C}_{2,\underline{\nu}} \end{pmatrix}, \quad \mathbb{C} = \begin{pmatrix} & \mathbb{C}_1 \\ -\frac{1}{\Delta t} \mathbb{C}_{2,\underline{\nu}} & \end{pmatrix}.$$

453 The Schur complement matrix \mathbb{S}^ε of dimension $2M_h \times 2M_h$ reads

$$454 \quad \mathbb{S}^\varepsilon = \begin{pmatrix} \mathbb{M}_\Gamma & \frac{\Delta t}{2} \mathbb{C}_{2,\underline{\nu}}^\top \mathbb{M}_2^{-1} \mathbb{C}_{2,\underline{\nu}} \\ -\frac{\Delta t}{2} \mathbb{C}_1^\top \mathbb{M}_1^{-1} \mathbb{C}_1 & \mathbb{M}_\Gamma \end{pmatrix}.$$

455 By re-numbering the unknown vector \vec{L} in the following way

$$456 \quad \vec{L} = \left((\vec{\Lambda}_v^{n+1})_1, (\vec{\Lambda}_s^{n+1})_1, \dots, (\vec{\Lambda}_v^{n+1})_{M_h}, (\vec{\Lambda}_s^{n+1})_{M_h} \right)^\top,$$

457 and applying the same operation in the Schur complement matrix, we observe that \mathbb{S}^ε is block diagonal with
458 each block defined as

$$459 \quad \left(\begin{array}{cc} (\mathbb{M}_\Gamma)_{II} & \frac{\Delta t}{2} \frac{(\mathbb{M}_\Gamma)_{II}^2}{(\mathbb{M}_2)_{\mathcal{V}_I^2(I)} \mathcal{V}_I^2(I)} \\ -\frac{\Delta t}{2} \frac{(\mathbb{M}_\Gamma)_{II}^2}{(\mathbb{M}_1)_{\mathcal{V}_I^1(I)} \mathcal{V}_I^1(I)} & (\mathbb{M}_\Gamma)_{II} \end{array} \right)_{I=1, \dots, M_h}. \quad (44)$$

460 Note that, to obtain (44), we used the fact that the interface mass matrix is diagonal and the expression
461 (35) of the transmission matrices. The determinant of each block is strictly positive, thus the computation
462 step (28) is well-posed. In practice, we store the inverse of each 2×2 block in an initializing step, and we
463 apply each inverse in parallel in order to solve (28) at each timestep, thus limiting the cost overhead of the
464 coupling procedures.

465 4.2.3. Coupling with PML formulations

466 Coupling with PML formulations can be handled in the same way. To illustrate this case, we suppose
467 that M_1 holds an acoustic formulation and M_2 an acoustic PML formulation such as the one given in (7).
468 We introduce the Lagrange multiplier

$$469 \quad \lambda = \underline{v}_1 \cdot \underline{\nu} = \underline{v}_2 \cdot \underline{\nu}.$$

470 The first relation can be expressed in terms of pressure variables as in (4.2.2). Assuming that

$$471 \quad \exists k \in \{1, \dots, d\} \quad \text{s.t.} \quad \underline{s}_k = \underline{\nu}, \quad (45)$$

472 where $\{\underline{s}_i\}_{i=1}^d$ is the set of splitting directions in (7), then we can express the second relation in the equivalent
473 form

$$474 \quad \lambda = (\underline{s}_k \otimes \underline{s}_k) \underline{v}_2 \cdot \underline{\nu}.$$

475 Note that the assumption (45) is satisfied in most practical cases. Using the relevant Green formula in both
476 systems we can introduce the Lagrange multiplier in both weak forms and retrieve the expression provided
477 in [35]. After time and space discretization, we obtain the following Schur complement matrix

$$478 \quad \mathbb{S}^\varepsilon = \varepsilon \mathbb{M}_\Gamma + \frac{\Delta t}{2} \mathbb{C}_1^\top \mathbb{M}_1^{-1} \mathbb{C}_1 + \frac{\Delta t}{2 + \tau_k \Delta t} \mathbb{C}_2^\top \mathbb{M}_2^{-1} \mathbb{C}_2,$$

479 where $\{\tau_i\}_{i=1}^d$ are the (constant) absorption coefficients in each splitting direction. Readers can refer to [35]
480 for the details of the numerical scheme, not recalled here for the sake of conciseness. Note that, as for the
481 previous case (40), we obtain, in the conform case, a lumped Schur complement matrix.

482 4.3. *N-domain problems*

483 So far we have considered two-domain problems exclusively and we have shown how, by using the
 484 natural trace of the volume discrete space, we can obtain lumped Schur complement matrices leading to very
 485 efficient coupling between sub-domains. Traditionally, this specific choice of discrete space for the Lagrange
 486 multipliers is prohibited since it may entail ill-posed Schur complement matrices, even in the conform case.
 487 To circumvent this difficulty, numerous possibilities of adequate discrete spaces for the multipliers have been
 488 proposed, e.g. see [29, 30, 47, 48]. In our work, we rely on an alternate method based upon a penalization
 489 strategy, suggested in [54]. This enables us to recover the invertibility of the Schur complement matrices,
 490 while keeping the computational performances of lumped matrices.

491 4.3.1. *Canonical example of an ill-posed lumped Schur complement matrix*

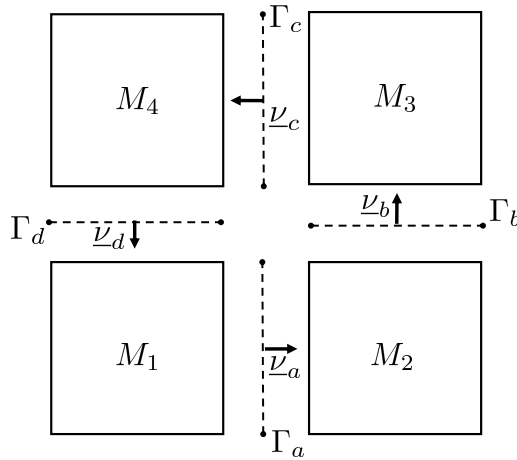


Figure 2: Illustration of the canonical example including a cross-point at the intersection of four interfaces.

492 To illustrate the loss of invertibility of the Schur complement matrix, we consider a canonical example
 493 of four sub-domains put together in a square-shape fashion, as depicted in Figure 2. In this example, the
 494 computational domain is composed of four macro-elements $\{M_1, M_2, M_3, M_4\}$ and four conform interfaces
 495 $\{\Gamma_a, \Gamma_b, \Gamma_c, \Gamma_d\}$. We denote by $\alpha = \{a, b, c, d\}$ a generic interface index and by $i, j \in \{1, 2, 3, 4\}$ two sub-
 496 domain indexes. To each interface Γ_α , we associate a normal vector field ν_α , oriented as depicted in Figure
 497 2. We assume that each sub-domain bears an acoustic formulation with identical mass densities and sound
 498 velocities. The Lagrange multiplier associated to each interface is defined as in (29). Additionally, we assume
 499 that the macro-elements are of the same size and that their corresponding meshes have the same number of
 500 elements and the same (isotropic) order of approximation. Similarly to (37), we can write in this context a
 501 system of the form of (25). The volume and interfaces unknowns are

$$502 \quad \vec{X} = (\vec{P}_1^{n+1} \dots \vec{P}_4^{n+1})^\top, \quad \vec{L} = (\vec{\Lambda}_a^{n+1} \dots \vec{\Lambda}_d^{n+1})^\top.$$

503 The matrix \mathbb{D} is the concatenation of four identical diagonal mass matrices \mathbb{M} , and \mathbb{D}^ε is the concatenation
 504 of four identical interface penalization matrices $\varepsilon \mathbb{M}_\Gamma$. We denote by $W_{h,\alpha}$ the discrete space for the Lagrange
 505 multipliers associated to the interface Γ_α , such that

$$506 \quad W_{h,\alpha} = \text{span}\{\mu_{\alpha,I}\}_{I=1}^{M_{h,\alpha}}, \quad M_{h,\alpha} = \dim(W_{h,\alpha}).$$

507 The transmission matrices are

$$508 \quad \mathbb{B} = \frac{\Delta t}{2} \begin{pmatrix} -\mathbb{C}_{1a} & & & \mathbb{C}_{1d} \\ \mathbb{C}_{2a} & -\mathbb{C}_{2b} & & \\ & \mathbb{C}_{3b} & -\mathbb{C}_{3c} & \\ & & \mathbb{C}_{4c} & -\mathbb{C}_{4d} \end{pmatrix}, \quad \mathbb{C} = -\mathbb{B}, \quad (46)$$

509 where $\forall i = 1, \dots, 4$ and $\forall \alpha \in \{a, b, c, d\}$

$$510 \quad \forall I = 1, \dots, N_{h,i}, \quad \forall J = 1, \dots, M_{h,\alpha} \quad (\mathbb{C}_{i\alpha})_{IJ} = \int_{\Gamma_\alpha} (\varphi_{i,I} | \Gamma) \mu_{\alpha,J} d\Gamma.$$

511 Note that in (46) the sign of the block matrices can be directly deduced from the chosen orientation of
512 the normal at each interface. The Schur complement matrix appearing in the computation step (28) is
513 symmetric and reads

$$514 \quad \mathbb{S}^\varepsilon = \mathbb{D}^\varepsilon + \frac{\Delta t^2}{4} \begin{pmatrix} \mathbb{S}_{12,a} & -\mathbb{C}_{2a}^\top \mathbb{M}_2^{-1} \mathbb{C}_{2b} & & -\mathbb{C}_{1a}^\top \mathbb{M}_1^{-1} \mathbb{C}_{1d} \\ \cdot & \mathbb{S}_{23,b} & -\mathbb{C}_{3b}^\top \mathbb{M}_3^{-1} \mathbb{C}_{3c} & \\ \cdot & \cdot & \mathbb{S}_{34,c} & -\mathbb{C}_{4c}^\top \mathbb{M}_4^{-1} \mathbb{C}_{4d} \\ \cdot & \cdot & \cdot & \mathbb{S}_{14,d} \end{pmatrix},$$

515 where we denote by (\cdot) the non-zero extra-diagonal blocks deduced from symmetry, and by $\mathbb{S}_{\alpha,ij}$ the diagonal
516 blocks such that

$$517 \quad \mathbb{S}_{ij,\alpha} = \mathbb{C}_{i\alpha}^\top \mathbb{M}_i^{-1} \mathbb{C}_{i\alpha} + \mathbb{C}_{j\alpha}^\top \mathbb{M}_j^{-1} \mathbb{C}_{j\alpha}, \quad \forall i = 1, \dots, 4, \quad \forall \alpha \in \{a, b, c, d\}. \quad (47)$$

518 For a fixed $i = 1, \dots, 4$ and $(\alpha, \beta) \in \{a, b, c, d\}^2$ such that $\alpha \neq \beta$, blocks of the form $\mathbb{C}_{i\alpha}^\top \mathbb{M}_i^{-1} \mathbb{C}_{i\beta}$ represent
519 the interactions between the Lagrange multipliers associated to the interfaces Γ_α and Γ_β through the macro-
520 element M_i . Hence, from the decomposition (35), one can see that these extra-diagonal blocks have only
521 one non-zero value at the cross-point. Assuming an identical numbering of the interfaces going from the
522 cross-point to the free extremity, we denote by

$$523 \quad I_\alpha^\times = 1, \quad I_\alpha^\circ = \llbracket 2; M_{h,\alpha} \rrbracket, \quad \forall \alpha \in \{a, b, c, d\},$$

524 the index of the node concerned with the cross-point and the rest of interface indexes in Γ_α respectively.
525 Following this notation, we propose the renumbering of the interface unknown

$$526 \quad \vec{L} = \left(\{\vec{L}_\alpha^\times\}_{\alpha=a}^d \quad \{\vec{L}_\alpha^\circ\}_{\alpha=a}^d \right)^\top, \quad (48)$$

527 so that the Schur complement matrix takes the form of

$$528 \quad \mathbb{S}^\varepsilon = \mathbb{D}^\varepsilon + \frac{\Delta t^2}{4} \begin{pmatrix} \mathbb{S}^\times & & & \\ (\mathbb{S}_{12,a})|_{I_a^\circ} & & & \\ & (\mathbb{S}_{23,b})|_{I_b^\circ} & & \\ & & (\mathbb{S}_{34,c})|_{I_c^\circ} & \\ & & & (\mathbb{S}_{14,d})|_{I_d^\circ} \end{pmatrix},$$

529 where \mathbb{S}^\times is the 4×4 cross-point matrix expressed as

$$530 \quad \mathbb{S}^\times = \begin{pmatrix} (\mathbb{S}_{12,a})|_{I_a^\times} & -(\mathbb{C}_{2a}^\top \mathbb{M}_2^{-1} \mathbb{C}_{2b})|_{I_a^\times I_b^\times} & & -(\mathbb{C}_{1a}^\top \mathbb{M}_1^{-1} \mathbb{C}_{1d})|_{I_a^\times I_d^\times} \\ \cdot & (\mathbb{S}_{23,b})|_{I_b^\times} & -(\mathbb{C}_{3b}^\top \mathbb{M}_3^{-1} \mathbb{C}_{3c})|_{I_b^\times I_c^\times} & \\ \cdot & \cdot & (\mathbb{S}_{34,c})|_{I_c^\times} & -(\mathbb{C}_{4c}^\top \mathbb{M}_4^{-1} \mathbb{C}_{4d})|_{I_c^\times I_d^\times} \\ \cdot & \cdot & \cdot & (\mathbb{S}_{14,d})|_{I_d^\times} \end{pmatrix}. \quad (49)$$

531 The expression of each extra-diagonal value of this cross-point matrix is

$$532 \quad (\mathbb{C}_{i\alpha}^\top \mathbb{M}_i^{-1} \mathbb{C}_{i\beta})|_{I_\alpha^\times I_\beta^\times} = \frac{(\mathbb{M}_{\Gamma_\alpha})|_{I_\alpha^\times I_\alpha^\times} (\mathbb{M}_{\Gamma_\beta})|_{I_\beta^\times I_\beta^\times}}{(\mathbb{M}_i)|_{\nu_{\Gamma_\alpha}^i(I_\alpha^\times) \nu_{\Gamma_\beta}^i(I_\beta^\times)}}, \quad \forall i = 1, \dots, 4, \quad \forall (\alpha, \beta) \in \{a, b, c, d\}^2, \quad \alpha \neq \beta.$$

533 In the specific case of our canonical example, the mass and interface mass matrices are identical. We denote
 534 by m^\times and m_Γ^\times their corresponding values at the cross-point index. After factorization, the cross-point
 535 matrix takes the simpler form of

$$536 \quad \mathbb{S}^\times = \frac{(m_\Gamma^\times)^2}{m^\times} \begin{pmatrix} 2 & -1 & & -1 \\ -1 & 2 & -1 & \\ & -1 & 2 & -1 \\ -1 & & -1 & 2 \end{pmatrix},$$

537 and has a non-empty kernel generated by the constant vectors

$$538 \quad \text{Ker}(\mathbb{S}^\times) = \text{span}\{(1 \ 1 \ 1 \ 1)^\top\}. \quad (50)$$

539 Thus, we can see the necessity of the penalization strategy since the Schur complement matrix \mathbb{S}^ε is ill-posed
 540 for $\varepsilon = 0$. It should be noted that the simple form (50) of the kernel of the cross-point matrix is linked to
 541 the specific parameters of our canonical configuration. In more generic cases, we expect this kernel to vary
 542 depending on the discretization patterns and the material properties of the macro-elements.

543 4.3.2. Schur complement matrix for N-domain problems

544 We consider a specific data-structure, referred to as the “skeleton” of the macro-mesh, regrouping the
 545 information of the incident faces (or edges) at every cross points, and the incident macro-elements at every
 546 interfaces. Using this “skeleton”, we can assemble the set of cross-point matrices, denoted by $\{\mathbb{S}_r^\times\}_{r=1}^{N_\times}$, where
 547 N_\times is the number of cross-points in the macro-mesh. If $d = 2$, the cross-points are restricted to points with
 548 multiple incident edges, while, if $d = 3$, the cross-points are spread across edges with multiple incident faces.
 549 The form of the cross-point matrices is similar to (49) with dimensions equal to the number of incident faces
 550 (or edges). We also assemble the Schur complement matrices set on the interior nodes. Let \mathcal{A} be the set
 551 of interface indexes, such that for any $\alpha \in \mathcal{A}$, the interface Γ_α is shared by two macro-elements $M_{i(\alpha)}$ and
 552 $M_{j(\alpha)}$. These interior Schur complement matrices are denoted by $\{\mathbb{S}_{i(\alpha)j(\alpha),\alpha}^\circ\}_{\alpha \in \mathcal{A}}$, with expressions deduced
 553 from (47). Finally, generalizing the previous renumbering (48), we can express the Schur complement matrix
 554 for N-domain problems in the general form of

$$555 \quad \mathbb{S}^\varepsilon = \begin{pmatrix} \{\mathbb{S}_r^\times\}_{r=1}^{N_\times} & \\ & \{\mathbb{S}_{i(\alpha)j(\alpha),\alpha}^\circ\}_{\alpha \in \mathcal{A}} \end{pmatrix}.$$

556 In practice, the interior Schur matrices are diagonal and every cross-point matrices are independent matrices
 557 of reasonable size so that we can compute and store their inverse in an initializing step. Applying their inverse
 558 at every timesteps is performed in parallel, leading to a limited cost overhead.

559 **Remark 2.** *The penalization procedure is a compromise between stability and consistency. In order to limit*
 560 *spurious consistency errors, we set $\varepsilon = \mathcal{O}(\Delta t^3)$ which is below the consistency errors of the numerical time*
 561 *schemes used for the time-discretization of the various formulations. Additionally, since only the cross-point*
 562 *matrices are ill-posed, we use an “inhomogeneous” penalization parameter set to zero at the interior nodes*
 563 *of the interfaces.*

564 5. Numerical simulation of immersed curved CFRP in 3D

565 As a numerical illustration, we propose to model the UT of a 4 mm thick immersed stratified composite
 566 material. The specimen is composed of 16 isotropic transverse plies of 235 μm thickness. The anisotropy
 567 direction of the plies, representing the fiber orientation, change from $\underline{f}_0 = (1 \ 0 \ 0)^\top$ to $\underline{f}_{90} = (0 \ 1 \ 0)^\top$
 568 successively from one ply to another. Between each ply and on the top face of the structure we consider thin
 569 intermediate epoxy layers of 15 μm represented as isotropic materials. Overall, the stratification is made of
 570 32 layers. Using Voigt notation to represent the constitutive law \mathcal{C}^* in (5), the material properties associated

	Density ($g \cdot cm^{-3}$)	Constitutive law coefficients (GPa)								
		C_{11}	C_{22}	C_{33}	C_{12}	C_{13}	C_{23}	C_{44}	C_{55}	C_{66}
Ply	1.6	143.2	15.8	15.8	7.5	7.5	8.2	3.8	7	7
Epoxy	1.23	7.6	7.6	7.6	4.4	4.4	4.4	1.6	1.6	1.6

Table 1: Material properties of the ply and epoxy layers.

571 to each ply and epoxy layers are presented in Table 1. The surrounding fluid is water with a mass density
572 of $\rho_w = 1.0 \text{ g} \cdot \text{cm}^{-3}$ and a sound velocity of $c_w = 1.483 \text{ mm} \cdot \mu\text{s}^{-1}$.

573 We assume that the structure is subject to a cylindrical deformation with respect to the y -axis with a
574 curvature center positioned at 10 mm below the specimen. We consider an incident pressure field p^{inc} as a
575 plane wave with a spatial Gaussian window,

$$576 \quad p^{\text{inc}}(\underline{x}, t) = f\left(\frac{(\underline{x} - \underline{x}_0) \cdot \underline{d}}{c_w} + t\right) \exp\left(-\frac{\|\underline{x} - \underline{x}_0\|^2}{\sigma_G}\right), \quad (51)$$

577 where $\underline{x}_0 = (0 \ 0 \ 3)^\top$ is the incident wave origin, $\underline{d} = (0 \ 0 \ -1)^\top$ is the direction of the propagation, the
578 standard deviation of the Gaussian window is set to $\sigma_G = 1.0$, and we define the excitation signal

$$579 \quad f(t) = \cos(2\pi F(t - t_0)) \exp\left(-\pi \frac{(t - t_0)^2}{\sigma_f}\right),$$

580 with $F = 3 \text{ MHz}$, $t_0 = 0.75 \mu\text{s}$ and $\sigma_f = 0.2$. In the following subsections, we use our numerical model to
581 compute the total field, i.e. the sum of the incident field and its interactions with the stratified specimen.
582 Due to the localized spatial support of the incident field, we truncate the numerical domain in the tangent
583 plane of the specimen so that its lengths are reduced to 6 mm in both directions. To avoid spurious
584 reflections, we surround the solid area with 1 mm solid PMLs. Above and below, we append two 1 mm
585 fluid sub-domains surrounded with additional 1 mm thick fluid PMLs. To sum up, prior to its cylindrical
586 deformation, the overall numerical domain has a size of $8 \times 8 \times 8 \text{ mm}^3$, and the disposition of the various
587 formulations are depicted in Figure 3. The incident field is evaluated at the upper interface between fluid
588 and fluid PML sub-domains.

589 **Remark 3.** In Appendix B we show how the transmission conditions are adequately taken into account
590 in the mortar element method. As a result, above the entry surface in the fluid PMLs, we solve the field
591 equations for the diffracted field and below for the total field. In practice, this enables us to restrict the
592 surrounding fluid sub-domains to a limited size while avoiding spurious interferences between the incident
593 and total fields.

594 5.1. Simulation in the case of a healthy curved CFRP

595 To start with we consider the case of a healthy specimen without internal flaw, as presented in Figure 3,
596 for which we have a total of 45 macro-elements. Note that, due to the cylindrical transformation, the
597 constitutive law varies locally and is expressed as in (5). In this case the local fiber orientation of each ply
598 is computed using (24), with $\tilde{\nu}$ being either \underline{f}_0 or \underline{f}_{90} .

599 In this configuration, the smallest wavelength is the one related to the fluid material, which is about
600 0.5 mm . Hence, in the tangent plane we perform a sub-discretization of each macro-element so that we can
601 insure two elements of order four per wavelength. In the thickness of the solid material, since the size of each
602 layer is smaller than half of the wavelength, we use a specific discretization pattern and use one element of
603 order three for the each ply and one element of order two for each epoxy layer. Note here the benefits of
604 the anisotropic orders in the finite element discretization, depicted in Section 2.2.

605 In Table 2 we gather the various characteristics of the simulation. We differentiate the number of nodes
606 in the final (sub-discretized) finite element space from the number of Degrees of Freedom (DoF). The latter is
607 obtained by taking into account the dimension of the unknown at each node, which is 1 for fluid, 3 for solid,

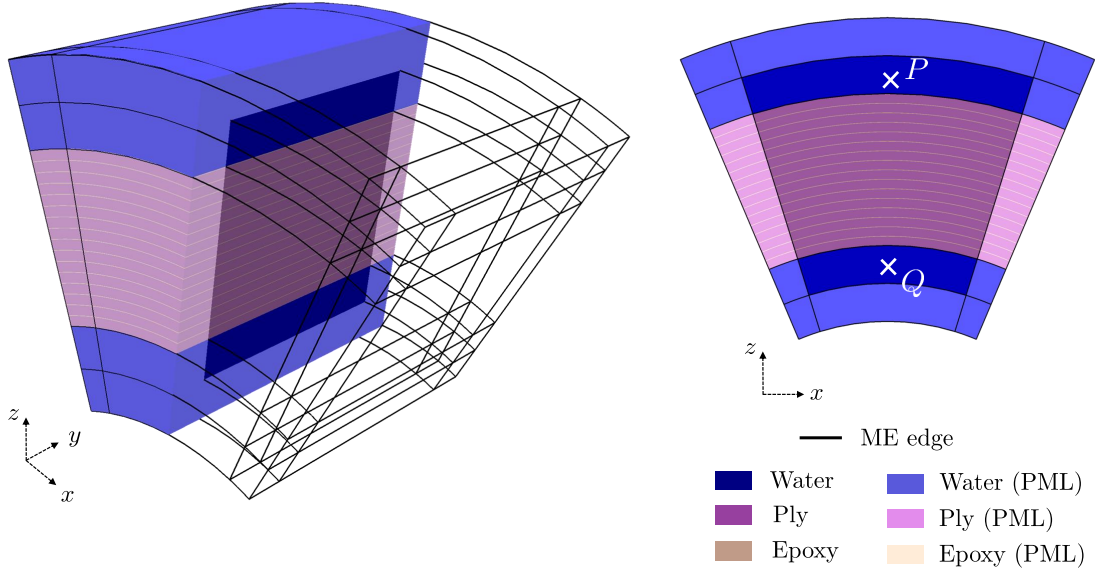


Figure 3: Numerical domain for the healthy immersed curved CFRP.

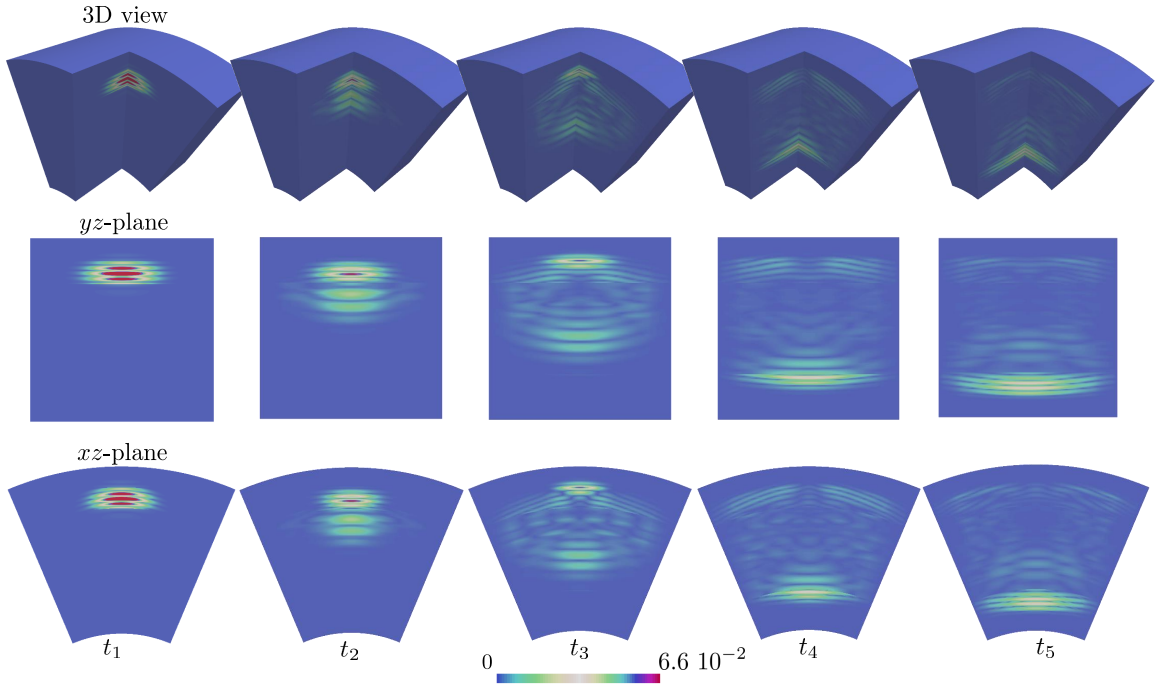


Figure 4: Snapshots of the absolute value of the pressure field in the fluid and fluid PML sub-domains, and the norm of the displacement in the solid and solid PML sub-domains. The snapshot times are $\{t_1, t_2, t_3, t_4, t_5\} = \{1.11, 1.66, 2.22, 2.77, 3.14\} \mu s$. For illustration purposes the pressure field is scaled up by 10%.

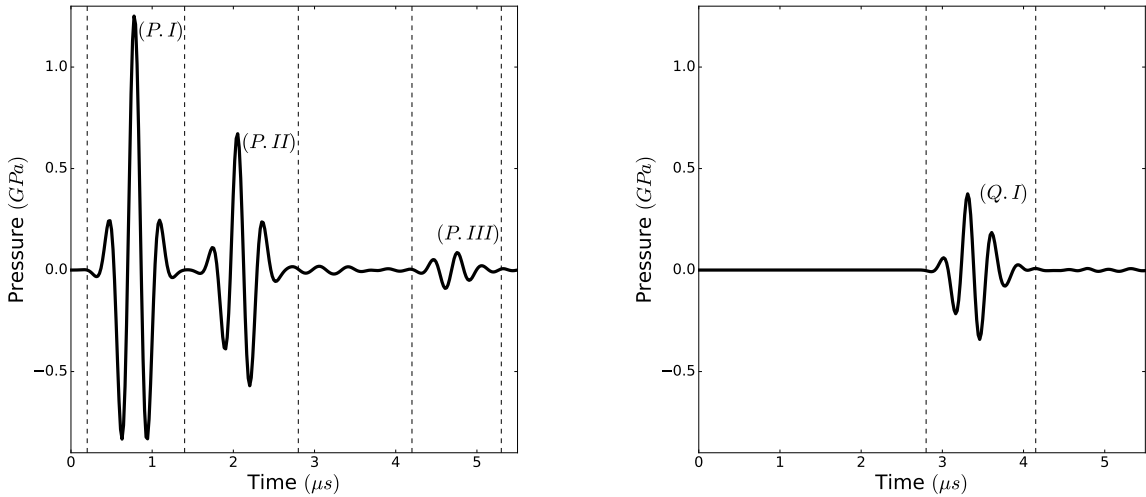
608 6 for fluid PML and 18 for solid PML sub-domains. It is important to emphasize that, even though we are
 609 facing a reasonably large number of DoFs within a locally varying anisotropic context, the memory footprint

610 of the overall numerical scheme is quite low (less than 1 *Gbytes*). Additionally, for reaching the maximal
611 time corresponding to the propagation of the ultrasonic beam through the structure, the computational time
612 is about ten minutes. These performances in both CPU time and memory load enable parametric variations
613 of this 3D configuration on a standard PC. The simulation was carried out on a bi-processor computer *Intel*
614 *Xeon Processors E5-2687W v2*, 2×8 cores at 3.40 *GHz*.

Nb. sub-domains				Nb. nodes				Nb. DoFs
Fluid	Fluid PML	Solid	Solid PML	Fluid	Fluid PML	Solid	Solid PML	
2	34	1	8	580 k	1665 k	1120 k	1045 k	32.7 M

Max. time (μs)	Timestep (μs)	Nb. steps	Mem. load (<i>bytes</i>)	CPU time (<i>min.</i>)
5.5	$1.85 \cdot 10^{-3}$	2973	790 M	10.53

Table 2: Characteristics of the simulation in the case of a healthy curved stratified specimen.



(a) Fluid pressure at point $P = (0, 0, 2.95)$.

(b) Fluid pressure at point $Q = (0, 0, -2.95)$.

Figure 5: Fluid pressure at two observation points above and below the immersed curved stratified material.

615 We present in Figure 4 the snapshots of the solution at various timesteps. In these snapshots, we
616 observe the propagation of the primary ultrasonic beam influenced by the anisotropy of the structure and
617 the generation of structural noise coming from the multiple reflections at the layer interfaces. Additionally,
618 we consider an observation point $P = (0, 0, 2.95)^T$ located above the structure, and an observation point
619 $Q = (0, 0, -2.95)^T$ located below the structure, see Figure (3). We plot the pressure field obtained at the
620 observation point P in Figure 5a. We see three main contributions: $(P.I)$ is the incident field defined in
621 (51); $(P.II)$ is the reflection of the incident field at the structure's upper boundary; $(P.III)$ is the backwall
622 echo. The signal between $(P.II)$ and $(P.III)$ is the phenomena referred to as the structural noise. In Figure
623 5b we plot the signal at point Q and we observe a main contribution $(Q.I)$ which is the transmitted wave
624 that traveled through the complete structure.

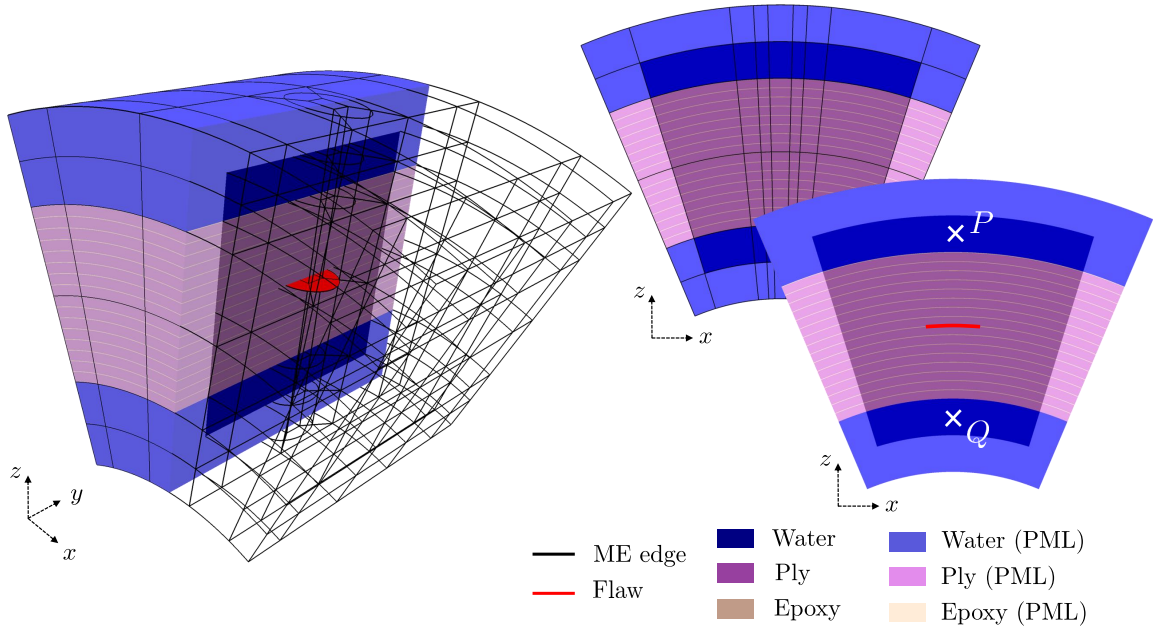


Figure 6: Representation of the numerical domain for the immersed curved CFRP with internal circular delamination of radius $r = 0.75$ mm.

5.2. Illustration of a variation of the configuration with a circular delamination flaw

Starting from the configuration described in the previous section, we introduce a circular delamination flaw located at the center of the structure. We add the necessary macro-elements so that the flaw geometry corresponds to an interface between two macro-elements. Between these two MEs, we discard the mortar element coupling scheme in order to incorporate the effect of the flaw in the numerical scheme. In terms of boundary conditions, this strategy is equivalent to representing the flaw by homogeneous Neumann boundary conditions. We show this configuration in Figure 6. In this more intricate case, the local anisotropy orientation of each ply is computed using (23), with \underline{v}^* being either \underline{f}_0 or \underline{f}_{90} .

In Table 3 we regroup the characteristics of the simulations performed as a toy example of parametric variation on the radius of the circular flaw $r \in \{0.5, 0.75, 1.00, 1.25, 1.50, 1.75\}$ mm. For this configuration we have 34 fluid, 98 fluid PML, 34 solid, and 32 solid PML sub-domains. Depending on the radius we adapt the geometries of the macro-elements, hence the number of nodes varies from one simulation to another. It should be noticed that, due to the inclusion of the flaw geometry and the conformity constraints that need to be satisfied, the final mesh incorporates elements of different sizes. As a consequence, the timestep has decreased compared to the healthy case, and the overall computation time has increased in order to reach an identical maximal time of $5.5 \mu s$.

We present in Figure 7 the snapshots of the solution at the same timesteps than in Figure 4. For this configuration we have fixed the radius of the circular delamination flaw to $r = 0.75$ mm. We observe the diffracted wave due to the presence of the flaw. This contribution is clearly seen in Figures 8a and 8b where we plot the pressure field at two observations points above and below the flawed structure. The contributions (P.I), (P.II) and (P.IV) are identical to the healthy case, and the contribution of the flaw is clearly identified in (P.III) and (Q.II). The backwall echo amplitude (P.IV) is lower than in the healthy case due to its interaction with a second transverse (and slower) flaw contribution.

In Figures 9a and 9b we plot the pressure at the same observation points for different values of the delamination radius $r \in \{0.5, 0.75, 1.00, 1.25\}$ mm. For readability, we restrict the plotting time window to the flaw responses exclusively. We observe that, as the flaw size increases, the primary contribution of the transmitted wave (P.III) increases and the primary contribution in the reflected wave (Q.I) decreases. Due

$r(mm)$	Nb. nodes				Nb. DoFs
	Fluid	Fluid PML	Solid	Solid PML	
0.50	873 k	2034 k	1704 k	1133 k	38.5 M
0.75	873 k	2034 k	1704 k	1133 k	38.5 M
1.00	1292 k	2439 k	2523 k	1119 k	43.6 M
1.25	1388 k	2549 k	2711 k	1133 k	45.2 M
1.50	1515 k	2662 k	2957 k	1119 k	46.5 M
1.75	1692 k	2853 k	3304 k	1133 k	49.1 M

$r(mm)$	Timestep (μs)	Nb. steps	Mem. load (<i>bytes</i>)	CPU time (<i>min.</i>)
0.50	$4.57 \cdot 10^{-4}$	12032	1001 M	47
0.75	$4.45 \cdot 10^{-4}$	11963	1017 M	48
1.00	$4.78 \cdot 10^{-4}$	11503	1214 M	60
1.25	$6.32 \cdot 10^{-4}$	8698	1266 M	48
1.50	$4.99 \cdot 10^{-4}$	11021	1321 M	65
1.75	$6.64 \cdot 10^{-4}$	8272	1410 M	52

Table 3: Characteristics of the simulations during the variation of the radius of the circular delamination flaw.

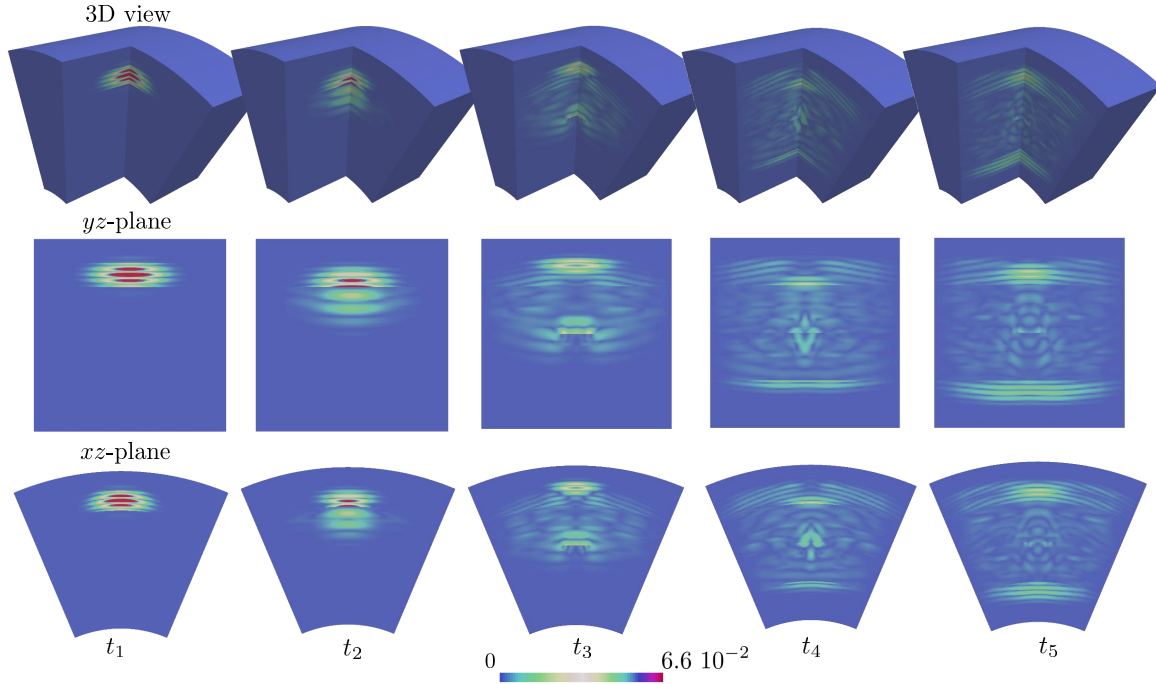
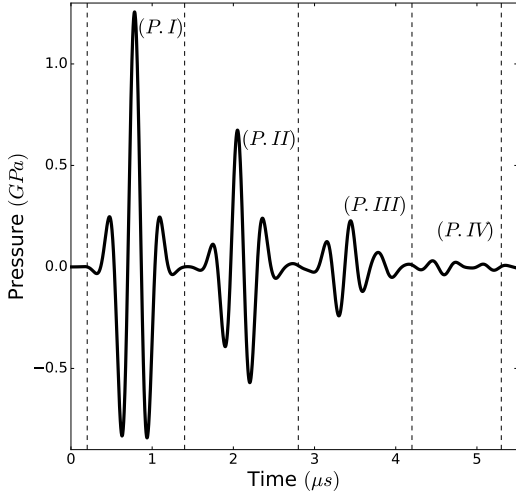


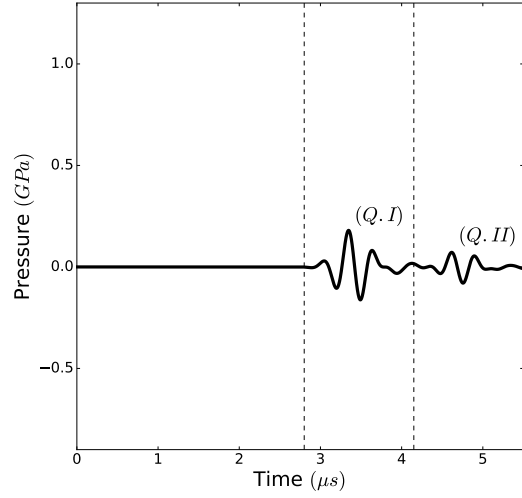
Figure 7: Snapshots of the solution in the presence of a circular delamination flaw with radius $r = 0.75 \text{ mm}$. The snapshot times are $\{t_1, t_2, t_3, t_4, t_5\} = \{1.11, 1.66, 2.22, 2.77, 3.14\} \mu s$. For illustration purposes the pressure field is scaled up by 10%.

652 to its interaction with the backwall echo, the contribution (*P.IV*) does not strictly follow this tendency and
653 it appears that the secondary transmitted wave is only slightly modified by the variation of the flaw.

654 **Remark 4.** *The approach used in this section can be extended to a class of hybrid methods. The incident*
655 *field proposed in (51) can be computed using a potentially more sophisticated model, e.g. a ray-based asymp-*
656 *tototic model. In a UT context, this strategy can be used to represent the propagation of the incident field,*

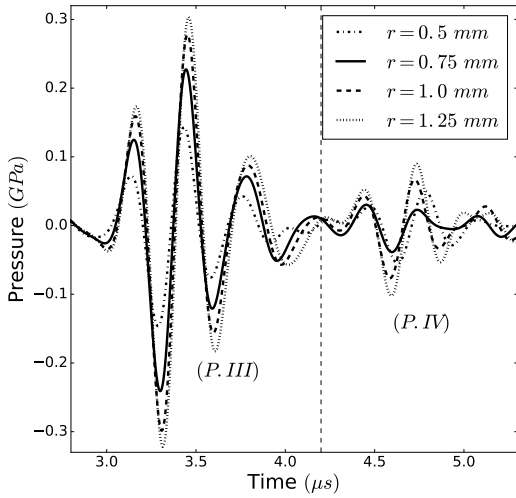


(a) Fluid pressure at point $P = (0, 0, 2.95)$.

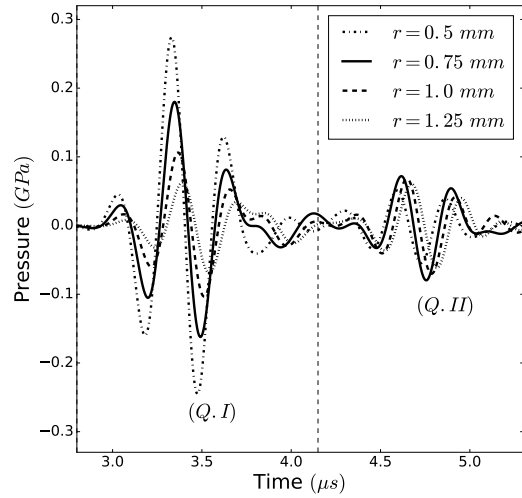


(b) Fluid pressure at point $Q = (0, 0, -2.95)$.

Figure 8: Fluid pressure at two observation points above and below the immersed curved stratified material with internal circular delamination of radius $r = 0.75$ mm.



(a) Fluid pressure at point $P = (0, 0, 2.95)$.



(b) Fluid pressure at point $Q = (0, 0, -2.95)$.

Figure 9: Variation of the fluid pressure at two observation points above and below the immersed curved stratified material depending on the radius of the internal circular delamination.

657 *without having to incorporate the complete propagation area in the numerical domain. When modeling real-*
 658 *istic UT configurations, it is also required of the forward solver to represent the electrical signal V_R captured*
 659 *by the receiving piezo-electric transducer. In this context, the pioneer works [1, 2, 55, 56] express the received*
 660 *electrical signal as a function of the so-called reciprocity quantity \mathcal{S} , namely*

$$V_R(t) = \mathcal{H} \star \mathcal{S}(t), \quad \forall t \in [0, T],$$

661

662 where T is the maximal time, $\cdot \star \cdot$ is the convolution in time domain, and \mathcal{H} is a convolution kernel repre-
 663 senting the sensitivity of the transducer acquisition chain. The reciprocity quantity \mathcal{S} reads

$$664 \quad \mathcal{S} = \int_{\Gamma} p^{\text{tot,E}} \star (\nabla p^{\text{inc,R}} \cdot \underline{n}) - p^{\text{inc,R}} \star (\nabla p^{\text{tot,E}} \cdot \underline{n}) \, d\Gamma,$$

665 where $p^{\text{tot,E}}$ is the total pressure field computed from the incident field generated by the emitting transducer,
 666 and $p^{\text{inc,R}}$ is a “virtually emitted” field from the receiving transducer. Note that Γ is a boundary surrounding
 667 the structure, or the flaw, and in the context of UT of composite materials we generally choose the entry
 668 surface. In practice, the numerical solver is used to compute the quantity $p^{\text{tot,E}}$, as presented in the numerical
 669 examples of this section. Hence, by appending the computation of the reciprocity signal to the finite element
 670 computations, our approach is compatible with this type of hybrid methods. For more details and illustrations
 671 of numerical results obtained using the development version of the CIVA software [12], readers may refer to
 672 prior communications [42, 43, 45, 57]

673 6. Conclusion and perspectives

674 In this work we have proposed a specific numerical solver for wave propagation modeling designed for
 675 UT configurations, and particularly suited for stratified composite materials. Our approach is focused on
 676 addressing, in an efficient and robust manner, parametric variations of the configuration, a fundamental pre-
 677 requisite for inversion loops or sensibility studies. To reach this objective, we have detailed a macro-element
 678 strategy which is based upon a decomposition of the configuration of interest. Each sub-domain, or macro-
 679 element, in this decomposition is associated to a specific wave propagation formulation, from which we derive
 680 a discrete propagator using the spectral finite element method. Making the most of the *a priori* information
 681 embedded in the scene decomposition, we are able to improve the performances of the standard finite
 682 element operations, in terms of memory footprint and computational load. Additionally, we have shown
 683 how we can extract from the macro-elements’ deformations important modeling components such as the
 684 anisotropy orientations or the splitting directions of PMLs, altogether in a lightweight and efficient fashion.
 685 The transmission conditions between each sub-domain are solved using the mortar element method, which
 686 is sufficiently flexible to take into account, in the same formalism, the various set of formulations arising in
 687 numerical UT modeling. In particular, we have depicted how, using the traces of the volume finite element
 688 spaces and a penalization strategy, one can apply a lumping integration technique on the Schur complement
 689 matrices in order to significantly reduce the cost of the coupling method, in the case of conform interfaces.
 690 To illustrate the efficiency of this approach we have proposed a 3D configuration of a curved composite
 691 materials, flawed with an internal circular delamination of varying radius.

692 A first and important perspective of this work would be to incorporate, within the macro-element strategy,
 693 the case of non-conform interfaces. Indeed, in numerous UT configurations, the wavelength ratio may vary
 694 strongly between different inclusions of materials. A typical example being the case of immersed specimen,
 695 as the one proposed in the numerical experiments of this communication. In these configurations, using
 696 the ability of the mortar element method to deal with independent sub-domain discretizations, we could
 697 significantly improve the efficiency of the global numerical scheme. However, the main issue, in this context,
 698 is to solve the interface linear system, which cannot be lumped, in an efficient manner. A potential lead to
 699 address this difficulty could be to rely on the conformity of the macro-mesh and to only consider “controled”
 700 non-conformities, where we have an *a priori* assumption on the refinement ratio between two adjacent
 701 discretizations. In addition to spatial non-conformities, we could also consider the class of methods that
 702 allows to include, within the same numerical scheme, different time-discretizations. We can mention, for
 703 instance, the local time-stepping method [58] or the locally implicit method [59]. Another possible way to
 704 improve the range of configurations efficiently handled by our approach could be to use the mortar element
 705 method to “implicitly” incorporate thin layers of materials, such as the epoxy layers in the presented
 706 numerical illustrations. We could, for instance, consider the standard spring-mass model [60] or a more
 707 complex asymptotic propagator associated to the thin layer, such as the one proposed in [61].

708 **Appendix A. Stability of a discrete coupled system**

709 *Appendix A.1. Discrete energy norm of second order schemes*

710 To start with, we recall a standard stability result, see [36] for more details. Considering the second
711 order fully discrete scheme (9), One can verify that the functional

$$712 \quad E^{n+\frac{1}{2}} = \frac{1}{2} \left\{ \frac{\vec{P}^{n+1} - \vec{P}^n}{\Delta t} \right\}^\top \mathbb{M} \left\{ \frac{\vec{P}^{n+1} - \vec{P}^n}{\Delta t} \right\} + \frac{1}{2} \vec{P}^{n+1} \mathbb{K} \vec{P}^n, \quad (\text{A.1})$$

713 satisfies a specific conservation property. Indeed, multiplying (9) by $\frac{1}{2\Delta t} \{ \vec{P}^{n+1} - \vec{P}^{n-1} \}$, we obtain

$$714 \quad \frac{1}{\Delta t} \{ E^{n+\frac{1}{2}} - E^{n-\frac{1}{2}} \} = 0.$$

715 To turn this conservation property into a stability result, we need to prove that the functional is positive for
716 any solution vector. While it is self-evident for the first term in (A.1), the second one needs some further
717 manipulations. Remarking that

$$718 \quad \vec{P}^{n+1} \mathbb{K} \vec{P}^n = \left\{ \frac{\vec{P}^{n+1} + \vec{P}^n}{2} \right\}^\top \mathbb{K} \left\{ \frac{\vec{P}^{n+1} + \vec{P}^n}{2} \right\} - \frac{\Delta t^2}{4} \left\{ \frac{\vec{P}^{n+1} - \vec{P}^n}{\Delta t} \right\}^\top \mathbb{K} \left\{ \frac{\vec{P}^{n+1} - \vec{P}^n}{\Delta t} \right\},$$

719 leads to the expression

$$720 \quad E^{n+\frac{1}{2}} = \frac{1}{2} \left\{ \frac{\vec{P}^{n+1} - \vec{P}^n}{\Delta t} \right\}^\top \left(\mathbb{M} - \frac{\Delta t^2}{4} \mathbb{K} \right) \left\{ \frac{\vec{P}^{n+1} - \vec{P}^n}{\Delta t} \right\} + \left\{ \frac{\vec{P}^{n+1} + \vec{P}^n}{2} \right\}^\top \mathbb{K} \left\{ \frac{\vec{P}^{n+1} + \vec{P}^n}{2} \right\}.$$

721 Hence, upon the CFL condition (10), the functional is positive and represents a discrete energy norm.

722 *Appendix A.2. Stability of the fluid-fluid coupling system*

We consider the case of the system (32). For $i = 1, 2$, let $\{E_i^{k+\frac{1}{2}}\}_{k \geq 0}$ be the discrete energy norm of
both sub-domains, defined as in (A.1). Multiplying each volume relation in (32) by $\frac{1}{2\Delta t} \{ \vec{P}_i^{n+1} - \vec{P}_i^{n-1} \}$,
and summing both equations yields

$$\begin{aligned} \sum_{i=1,2} \frac{1}{\Delta t} \{ E_i^{n+\frac{1}{2}} - E_i^{n-\frac{1}{2}} \} - \left\{ \frac{\vec{P}_1^{n+1} - \vec{P}_1^{n-1}}{2\Delta t} \right\}^\top \mathbb{C}_1 \left\{ \frac{\vec{\Lambda}^{n+1} - \vec{\Lambda}^n}{2\Delta t} \right\} \\ + \left\{ \frac{\vec{P}_2^{n+1} - \vec{P}_2^{n-1}}{2\Delta t} \right\}^\top \mathbb{C}_2 \left\{ \frac{\vec{\Lambda}^{n+1} - \vec{\Lambda}^n}{2\Delta t} \right\} = 0. \end{aligned}$$

723 Using the discrete form of the continuity relation in (32), we obtain

$$724 \quad \sum_{i=1,2} \frac{1}{\Delta t} \{ E_i^{n+\frac{1}{2}} - E_i^{n-\frac{1}{2}} \} + \varepsilon \left\{ \frac{\vec{\Lambda}^{n+1} - \vec{\Lambda}^n}{2\Delta t} \right\}^\top \mathbb{M}_\Gamma \left\{ \frac{\vec{\Lambda}^{n+1} - 2\vec{\Lambda}^n - \vec{\Lambda}^{n-1}}{\Delta t^2} \right\} = 0.$$

725 Introducing the positive functional

$$726 \quad E_\Gamma^{n+\frac{1}{2}} = \frac{1}{2} \left\{ \frac{\vec{\Lambda}^{n+1} - \vec{\Lambda}^n}{\Delta t} \right\}^\top \mathbb{M}_\Gamma \left\{ \frac{\vec{\Lambda}^{n+1} - \vec{\Lambda}^n}{\Delta t} \right\},$$

727 we finally get the following conservation property for the coupling system (32)

$$728 \quad \sum_{i=1,2} \frac{1}{\Delta t} \{ E_i^{n+\frac{1}{2}} - E_i^{n-\frac{1}{2}} \} + \frac{\varepsilon}{\Delta t} \{ E_\Gamma^{n+\frac{1}{2}} - E_\Gamma^{n-\frac{1}{2}} \} = 0. \quad (\text{A.2})$$

729 Therefore, assuming that the timestep satisfies the inequality (36), we can interpret the relation (A.2) as a
730 stability result, valid for any $\varepsilon \geq 0$.

731 **Appendix B. Diffracted field - total field coupling system**

732 In some cases, it is important to be able to introduce in the numerical domain an externally computed
 733 incident field. As detailed in Remark 4, a typical example is the case of hybrid coupling, where the global
 734 incident field p^{inc} is computed using a ray-based asymptotic model, and the total field is obtained in a specific
 735 area using a numerical solver. In practice, the mortar element method provides us with a convenient way
 736 to solve this transmission problem. To give an illustration, let us consider a two-domain acoustic problem
 737 such as the one presented in Section 4.2.1. For simplicity, we assume that $\rho_1 = \rho_2 = \rho$ and $c_1 = c_2 = c$. The
 738 second domain bears a total field formulation, the incident field is introduced at the interface between the
 739 two sub-domains, and the first domain is written in terms of diffracted field. The main step is to write the
 740 relations (29) in terms of total fluxes

$$741 \quad \frac{\partial \lambda}{\partial t} = \frac{1}{\rho} \nabla p_1^{\text{diff}} \cdot \underline{\nu} + \tau^{\text{inc}} = \frac{1}{\rho_2} \nabla p_2^{\text{tot}} \cdot \underline{\nu}, \quad \text{with } \tau^{\text{inc}} = \frac{1}{\rho} \nabla p^{\text{inc}} \cdot \underline{\nu},$$

742 and the weak continuity relation in (30) in terms of total traces

$$743 \quad \frac{d}{dt} \int_{\Gamma} (p_1^{\text{diff}} - p_2^{\text{tot}}) \lambda^* d\Gamma + \varepsilon \frac{d^2}{dt^2} \int_{\Gamma} \lambda \lambda^* d\Gamma = - \frac{d}{dt} \int_{\Gamma} p^{\text{inc}} \lambda^* d\Gamma.$$

744 Hence, provided that the outside model is able to compute the input data $(p^{\text{inc}}, \tau^{\text{inc}})$, we can incorporate
 745 them adequately as right hand sides in the coupled system (30).

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